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KAMAN AVIDYNE BURLINGTON MA

TRAP-ML-A TWO DIMENSIONAL THERMAL RESPONSE CODE TAILORED FOR TH--ETC(U)

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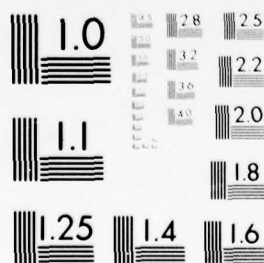
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MICROCOPY RESOLUTION TEST CHART
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TRAP-ML-A TWO-DIMENSIONAL THERMAL
RESPONSE CODE TAILORED FOR THE
DEFENSE NUCLEAR AGENCY TRI-SERVICE
THERMAL RADIATION TEST FACILITY

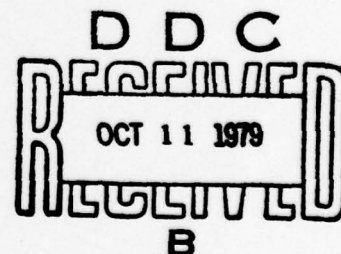
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30 November 1978

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20. ABSTRACT (Continue on reverse side if necessary and identify by block number) TRAP-ML is a digital computer code for calculating the two-dimensional thermal response of materials exposed to thermal radiation and wind tunnel cooling. In particular, the code can simulate the Defense Nuclear Agency Tri-Service Thermal Radiation Test Facility located at the Air Force Mate- rials Laboratory, WPAFB, which, in turn, can simulate an in-flight nuclear encounter. (over)		

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20. ABSTRACT (Continued)

The program is capable of analyzing multi-material, multi-layer samples exposed to either quartz-lamp radiation or a user-designated thermal source (as a function of both wavelength and time), with optional convective (air stream) cooling. Specific heats and thermal conductivities of the materials can be specified, in general, as functions of both temperature and radiation wavelength. 

For output, the code computes the heat flow and temperature distribution throughout the finite model, and will optionally plot the temperature-time histories at layer boundaries and at both surfaces of the specimen.

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PREFACE

The work was performed for the Defense Nuclear Agency (DNA) under Contract Number DNA001-78-C-0057 by Kaman Avidyne, Burlington, Mass., a division of Kaman Sciences Corporation. Captain J. M. Rafferty of DNA, and Major J. Hurst and Mr. G. Schmitt of the Air Force Materials Laboratory (AFML) at Wright-Patterson Air Force Base, Dayton, Ohio, were technical monitors. Mr. William Lee was project leader of the work performed in the Structural Mechanics Section of Kaman Avidyne headed by Mr. Emanuel S. Criscione, with technical contributions by Dr. J. R. Ruetenik and Dr. N. P. Hobbs.

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Conversion factors for U.S. customary
to metric (SI) units of measurement.

To Convert From	To	Multiply By
angstrom	meters (m)	1.000 000 X E -10
atmosphere (normal)	kilo pascal (kPa)	1.013 25 X E +2
bar	kilo pascal (kPa)	1.000 000 X E +2
barn	meter ² (m ²)	1.000 000 X E -28
British thermal unit (thermochemical)	joule (J)	1.054 350 X E +3
calorie (thermochemical)	joule (J)	4.184 000
cal (thermochemical)/cm ²	mega joule/m ² (MJ/m ²)	4.184 000 X E -2
curie	giga becquerel (GBq)*	3.700 000 X E +1
degree (angle)	radian (rad)	1.745 329 X E -2
degree Fahrenheit	degree kelvin (K)	$t_K = (t_F + 459.67)/1.8$
electron volt	joule (J)	1.602 19 X E -19
erg	joule (J)	1.000 000 X E -7
erg/second	watt (W)	1.000 000 X E -7
foot	meter (m)	3.048 000 X E -1
foot-pound-force	joule (J)	1.355 818
gallon (U.S. liquid)	meter ³ (m ³)	3.785 412 X E -3
inch	meter (m)	2.540 000 X E -2
jerk	joule (J)	1.000 000 X E +9
joule/kilogram (J/kg) (radiation dose absorbed)	Gray (Gy)**	1.000 000
kilotons	terajoules	4.183
kip (1000 lbf)	newton (N)	4.448 222 X E +3
kip/inch ² (ksi)	kilo pascal (kPa)	6.894 757 X E +3
ktap	newton-second/m ² (N-s/m ²)	1.000 000 X E +2
micron	meter (m)	1.000 000 X E -6
mil	meter (m)	2.540 000 X E -5
mile (international)	meter (m)	1.609 344 X E +3
ounce	kilogram (kg)	2.834 952 X E -2
pound-force (lbf avoirdupois)	newton (N)	4.448 222
pound-force inch	newton-meter (N-m)	1.129 848 X E -1
pound-force/inch	newton/meter (N/m)	1.751 268 X E +2
pound-force/foot ²	kilo pascal (kPa)	4.788 026 X E -2
pound-force/inch ² (psi)	kilo pascal (kPa)	6.894 757
pound-mass (lbm avoirdupois)	kilogram (kg)	4.535 924 X E -1
pound-mass-foot ² (moment of inertia)	kilogram-meter ² (kg-m ²)	4.214 011 X E -2
pound-mass/foot ³	kilogram/meter ³ (kg/m ³)	1.601 846 X E +1
rad (radiation dose absorbed)	Gray (Gy)**	1.000 000 X E -2
roentgen	coulomb/kilogram (C/kg)	2.579 760 X E -4
shake	second (s)	1.000 000 X E -8
slug	kilogram (kg)	1.459 390 X E +1
torr (mm Hg, 0° C)	kilo pascal (kPa)	1.333 22 X E -1

*The becquerel (Bq) is the SI unit of radioactivity; 1 Bq = 1 event/s.

**The Gray (Gy) is the SI unit of absorbed radiation.

A more complete listing of conversions may be found in "Metric Practice Guide E 380-74," American Society for Testing and Materials.

TABLE OF CONTENTS

<u>Section</u>	<u>Page</u>
1 INTRODUCTION - - - - -	7
2 HEAT TRANSFER MECHANISMS - - - - -	11
2-1 RADIATION SOURCE - - - - -	11
2-2 RERADIATION - - - - -	16
2-3 CONDUCTION - - - - -	16
2-4 CONVECTION - - - - -	18
2-5 TEMPERATURE CHANGE - - - - -	20
3 COMPUTER PROGRAM DESCRIPTION - - - - -	22
3-1 PROGRAM OPERATION - - - - -	22
3-2 PROGRAM INPUT - - - - -	25
3-3 PROGRAM OUTPUT - - - - -	39
4 SAMPLE PROBLEM AND COMPARISON WITH EXPERIMENT- - - - -	44
REFERENCES - - - - -	48
APPENDIX A - - - - -	49
APPENDIX B - - - - -	75

LIST OF ILLUSTRATIONS

<u>Figure</u>		<u>Page</u>
1	Spectral intensity model for nominal radiation source - - - - -	12
2	Time-wise exposure model of nominal radiation source - - - - -	15
3	General two-dimensional cross-sectional model indicating elements, layers and segments - - - - -	17
4	Thermal conductivity versus temperature - - - - -	37
5	Specific heat versus temperature - - - - -	38
6	Time-history output for sample problem - - - - -	41
7	Sample problem input data - - - - -	45
8	Computer-generated sample plot - - - - -	46
9	Comparison of TRAP-ML predictions to experiment - - -	47

LIST OF TABLES

<u>Table</u>	<u>Page</u>
1 Conversion of units (English to metric) - - - - -	10
2 Wavelengths used to determine absorbed flux - - - - -	13
3 Typical job control sequence - - - - -	23
4 TRAP-ML and APLOT routines - - - - -	24
5 Description of variables which determine program dimensions - - - - -	26
6 TRAP-ML input - - - - -	27
7 Material properties in TRAP-ML - - - - -	36
8 Plot symbols used to identify plot curves - - - - -	42
9 Error codes from INT1Z - - - - -	43

SECTION 1

INTRODUCTION

TRAP-ML is the result of a dual effort to (1) simplify the TRAP computer code (Reference 1) for studying radiation effects on aircraft components and (2) provide an analytical predictive and correlative tool for use in the Defense Nuclear Agency (DNA) Tri-Service Thermal Radiation Test Facility located at the Air Force Materials Laboratory, WPAFB, Dayton, Ohio. The appendage "ML" on the code name indicates the special applicability of the code to the Materials Laboratory.

The DNA thermal test facility, described in detail in References 2, 3 and 4, includes a test setup consisting of a quartz lamp bank for thermal radiation and a wind tunnel providing convective cooling on material samples approximately four inches square. The samples are mounted flush with the inside surface of the wind tunnel, facing a window in the opposite tunnel wall. The sample is irradiated through this window, simulating the environment of an aircraft exposed to nuclear radiation while in flight.

The TRAP code (Thermal Response of Aircraft to Nuclear Radiation) includes the material modelling and heat transfer capabilities required for the test facility application. The solution method is based on a finite representation of the cross section by thermally thin "elements." One or more elements make up a "layer," characterized by uniform material properties and thickness, and several layers are possible. The layers make up a "segment"; several segments are allowed, introducing the possibility of two-dimensional heat flow. The heat transfer mechanisms in TRAP include radiation, convection, and conduction.

The problem with the original TRAP is that it is too general for the test facility application. For example, it also includes a nuclear source generator, atmospheric transmission and ground reflection factors, and a vulnerability assessment including an iteration on slant

range. Also, it is designed to analyze large complex aircraft structures - not laboratory samples. Clearly, these capabilities are not needed for the test facility. Nor are the stress-strain calculations needed, as the mechanical boundary conditions are not prescribed in the laboratory setup.

Certain other capabilities were required in TRAP-ML, however, which are not in the original TRAP. The quartz lamp radiation source was modelled as a function of wavelength and time and included in the program. Provision was also made for two other source options, including a point-by-point option.

While the test facility setup can be treated as a one-dimensional thermal problem, the capability to analyze two-dimensional heat flow remains. For example, the airstream does tend to "heat up" downstream, introducing a nonuniform axial temperature field. So, rather than eliminate the 2-D capability, it was retained for possible future applications.

Also, while convection and reradiation are still important, these heat transfer processes were modified slightly to better simulate the test facility. Input and output were also modified to enable the user to operate the code with less than a 60,000₈ core requirement and receive temperature-versus-time plots at each surface and layer boundary of the material sample.

TRAP-ML, then, allows the user to simulate a multi-layer, multi-material sample subjected to a general radiation source (a 6000 watt tungsten lamp in particular), complete with two-dimensional heat flow, convective cooling (if needed), and reradiation, with both the source and material absorptivity specified, in general, as functions of wavelength.

With this capability, the pretest calculations can be performed which, in many cases, may reduce the number of test exposures required. And where material properties are uncertain, post-test correlation with tests involving these materials may lead to evaluations of the unknown properties.

Section 2 describes in more detail the heat transfer processes in TRAP-ML, and Section 3 documents the computer program. Section 4 includes a sample problem which is compared with a test exposure of a .032 inch aluminum sample coated with 3 mils of white polyurethane. This limited correlation indicates excellent agreement. Finally, Table 1 includes a conversion table of units (English to Metric), since the TRAP computer code (and hence, TRAP-ML) was originally formulated in the English system of units, including input and output. Program listings make up Appendices A and B.

Table 1. Conversion of units (English to metric).

Value Expressed in English Units	Multiplied By	Yields Value Expressed In Metric Units
Inch (in)	2.54	Centimeters (cm)
British Thermal Unit (BTU)	252.0	Calories (cal)
BTU/in ²	39.03	Cal/cm ²
Temperature, °R	0.5555	Temperature, °K
BTU/in ² /°R	70.25	Cal/cm ² /°K
BTU/pound mass (BTU/lb-m)	0.5555	Cal/gram
BTU/lb-m/°R	1.0	Cal/gram/°K

SECTION 2

HEAT TRANSFER MECHANISMS

2-1 RADIATION SOURCE

The primary radiation source included in the program is the quartz lamp bank consisting of 6000 watt tungsten filament bulbs. The spectral distribution and a general time history are included in a non-dimensional manner, such that the user need only specify the peak flux level to be achieved and the time of exposure before the lamps are shut off. Provision is also made for other source characteristics, or even very simple sources such as a step-on and step-off time history.

The nominal spectral distribution is based on spectrometer measurements and the point-by-point model is indicated in Figure 1. While there is some absorption of the radiation by the atmosphere, it is small since the lamps are normally located within about 2 inches of the thermal sample, and is, therefore, neglected. The distribution is considered invariant with time, although it is thought the peak intensity may shift to a higher wavelength when cool. While this is ignored in the nominal source option, the capability exists of specifying the source as a function of both wavelength and time.

The program numerically integrates the distribution between 0.2 and 5.5 microns to determine the total flux, then normalizes the entire distribution to that value as a function of 14 discrete wavelengths. If wavelengths other than the nominal 14 are specified on input, the program linearly interpolates between the inputted values to determine values for the 14 basic wavelengths. These fourteen wavelengths and their associated band widths are indicated in Table 2. When material absorptivity is specified as a function of wavelength, these 14 bands are used to numerically integrate the total absorbed radiation between 0.2 and 5.5 microns. Since all the incident radiation

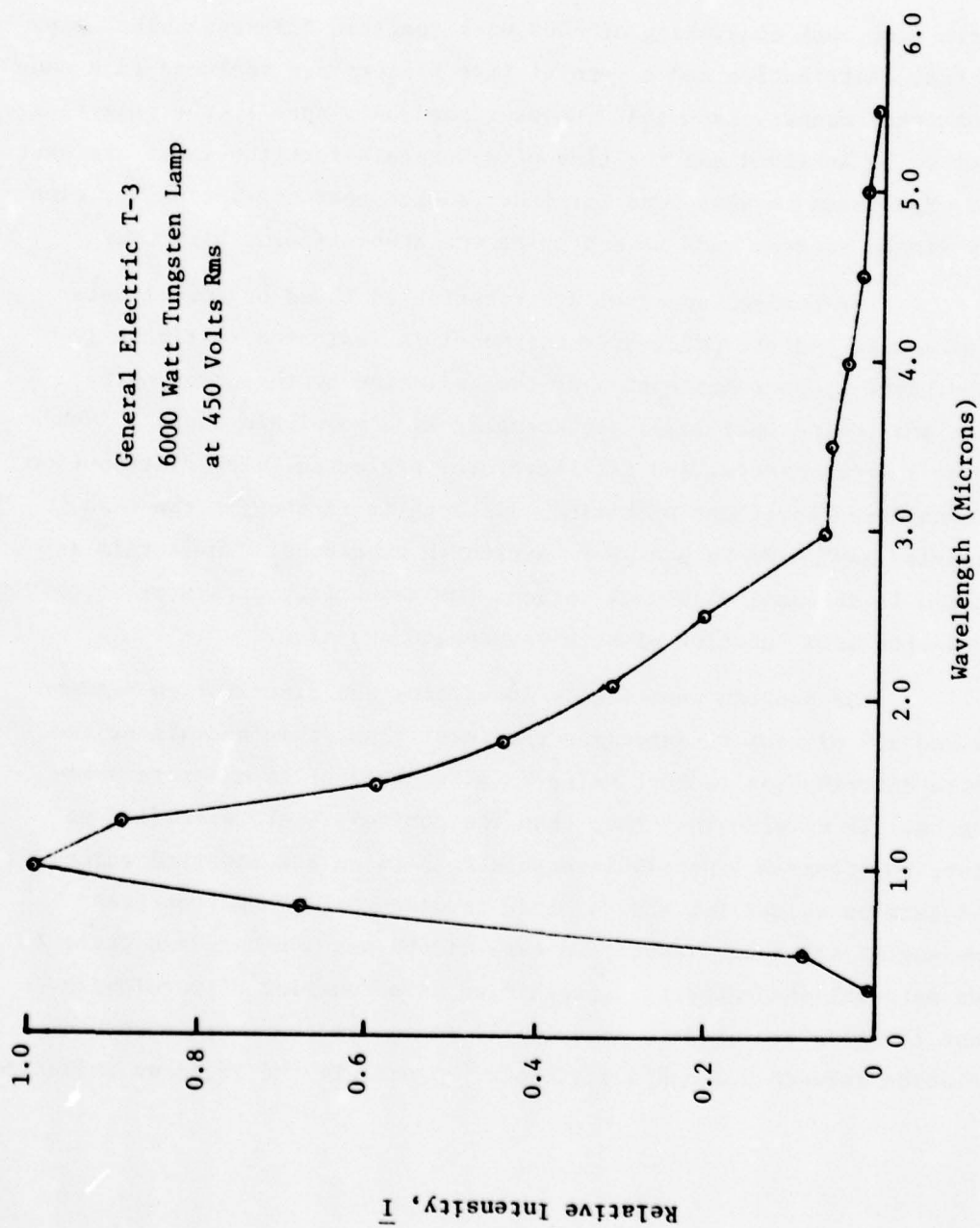


Figure 1. Spectral intensity model for nominal radiation source.

Table 2. Wavelengths used to determine absorbed flux.

Wavelength Number	Nominal Wavelength (Microns)	Wavelength Band (Microns)
1	0.288	0.2 - 0.375
2	0.5	0.375 - 0.625
3	0.75	0.625 - 0.875
4	1.0	0.875 - 1.125
5	1.25	1.125 - 1.375
6	1.5	1.375 - 1.625
7	1.75	1.625 - 1.875
8	2.0625	1.875 - 2.25
9	2.5	2.25 - 2.75
10	3.0	2.75 - 3.25
11	3.5	3.25 - 3.75
12	4.0	3.75 - 4.25
13	4.5	4.25 - 4.75
14	5.0	4.75 - 5.25

is either absorbed or reflected back into space, the incident absorbed relative intensity can be given by

$$\bar{I}^{abs} = \frac{\int_{0.2}^{5.5} \alpha(\lambda) \bar{I}(\lambda) d\lambda}{\int_{0.2}^{5.5} \bar{I}(\lambda) d\lambda} \quad (1)$$

where α represents the fractional absorptivity ($0 \leq \alpha \leq 1$) and λ is the wavelength.

Figure 2 indicates the nominal time-wise radiation history, where the flux has been normalized to an arbitrary value. The time history is divided into two domains:

- 1) The "rise" domain includes a very sharp rise in flux and then reaches a nearly steady-state value for large times. After 3.87 seconds the program assumes a constant flux.
- 2) The decay phase commences when the lamps are turned off and has a rapid decrease in flux until it finally is assumed to be zero at 2.85 seconds after the exposure has ended.

Regardless of when the lamps are turned off, the entire curve is normalized to a value of peak flux (q_{max}) which is specified on input. And the decay phase begins exactly with the time specified on input. The incident absorbed flux can then be given by

$$q^{abs}(t) = q_{max} \cdot \bar{I}^{abs} \cdot \bar{F}(t) \quad (2)$$

where \bar{F} is the relative flux defined in Figure 2 and t represents time.

The user can also specify a simple step-on and step-off time history or a general point-by-point description. Linear interpolation is used between tabulated values. Even these distributions are normalized to the largest tabulated value so that the intensity of the radiation can be varied with a single input parameter. The entire specimen is assumed to be irradiated uniformly.

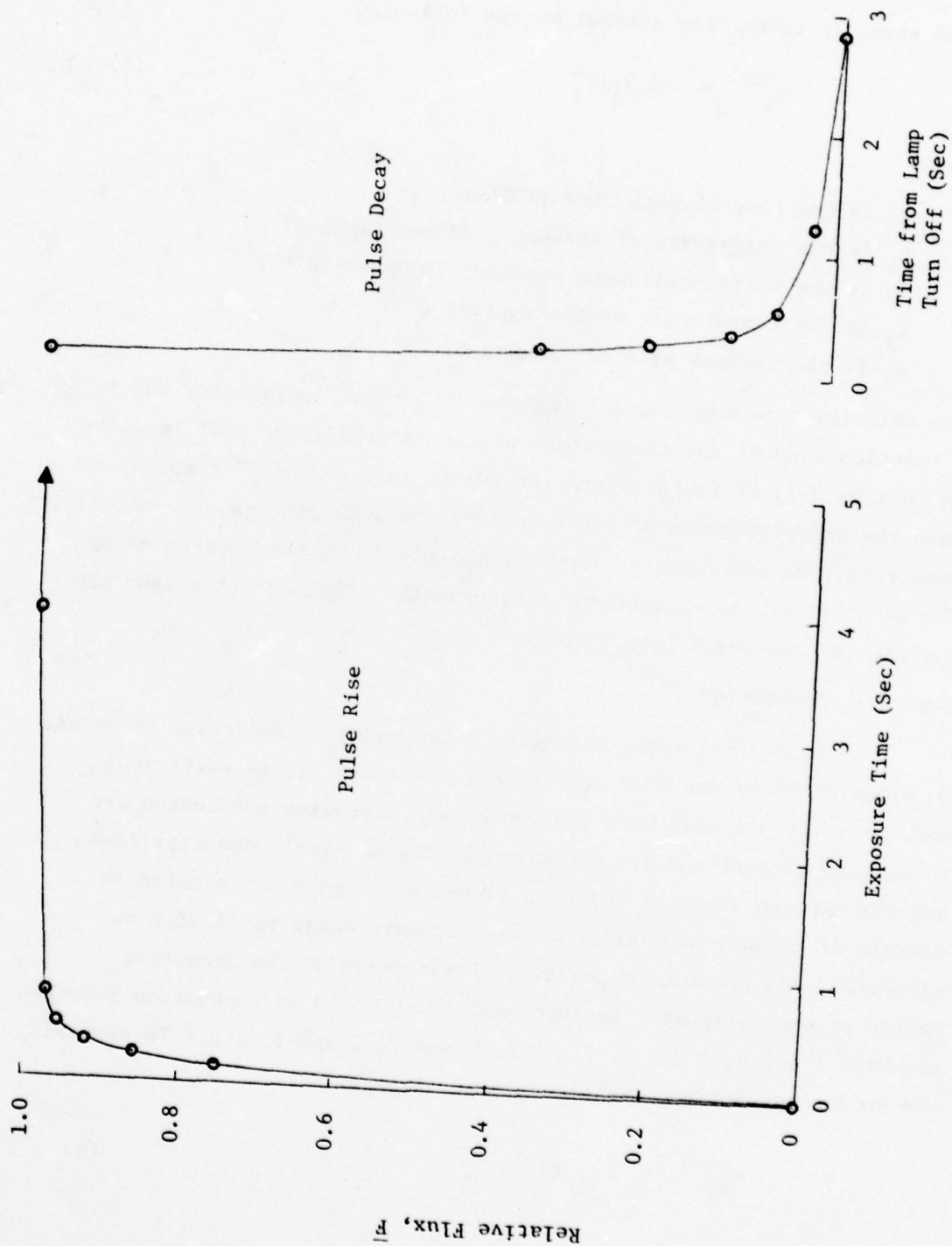


Figure 2. Time-wise exposure model of nominal radiation source.

2-2 RERADIATION

Reradiation of energy from each surface of the sample back into space is taken into account by the following:

$$q_i^{\text{rad}} = -A_i \epsilon_i \sigma T_i^4 \quad (3)$$

where

q_i is the rate of heat flow (BTU/sec)

ϵ_i is the emissivity of surface i (dimensionless)

σ is the Stefan-Boltzmann constant (BTU/sec-in²-°R⁴)

T_i is the temperature of the surface (°R)

A_i is the surface area of element i (in²)

The emissivity is considered "hemispherical total" emissivity and is a function only of the temperature of the material. As will be noted in Section 3-1, if the absorptivity (α) is independent of temperature, then the emissivity can be taken as $1-\alpha$. But, in general, the emissivity must either be specified on input or calculated by the program using the absorptivity as a function of wavelength. The heat flow from the surface is considered lost forever.

2-3 CONDUCTION

Conduction among elements in the model is described in detail on pages 86-92 of the TRAP report (Reference 1). It is sufficient here to point out that both lengthwise and depthwise conduction are taken into account and are functions of the material conductivities and the exposed areas of adjacent elements. Figure 3 indicates an example of a model consisting of two segments and a total of nine elements. The element "nodes" are always taken at the geometric center of each element. In this case there would be conduction between elements 1 and 6, 2 and 1, 2 and 6, 2 and 7, 2 and 3, etc. In general, the conduction between two elements is given by

$$q_{ij}^{\text{cond}} = \bar{K}_{ij} (T_i - T_j) \quad (4)$$

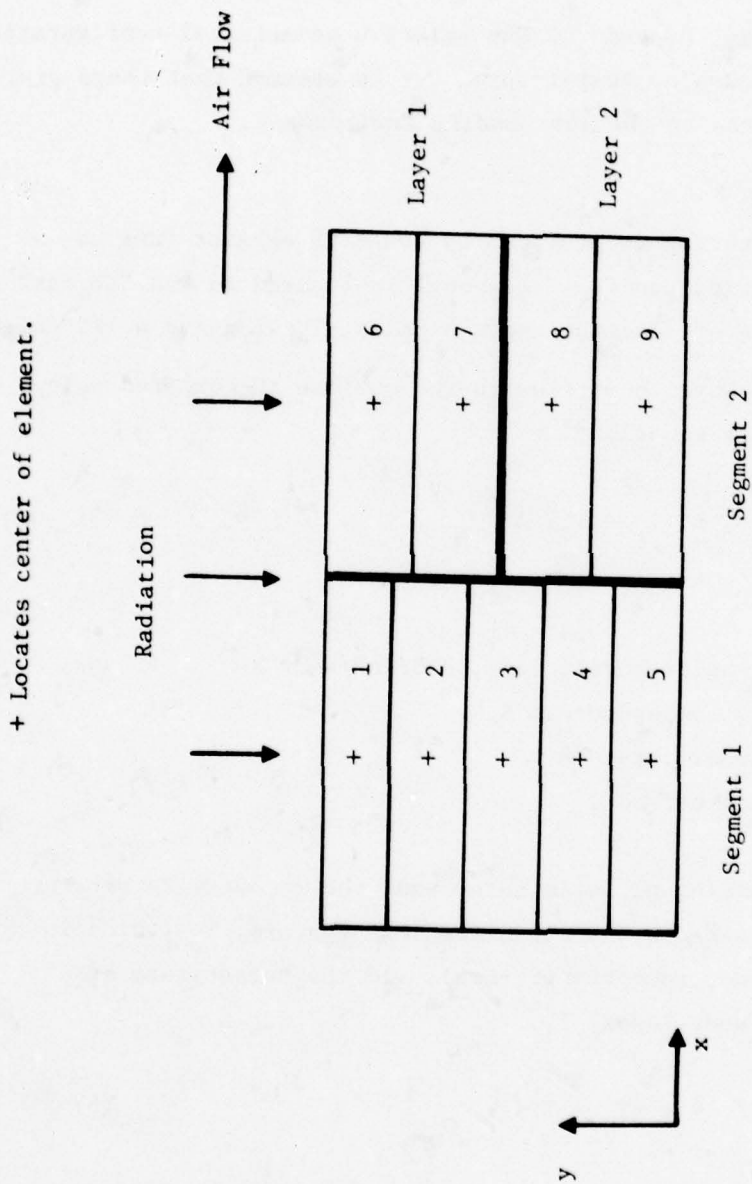


Figure 3. General two-dimensional cross-sectional model indicating elements, layers and segments.

where

\bar{K}_{ij} is a conductivity factor between elements i and j, BTU/sec-°R
T is temperature, °R

As indicated earlier, \bar{K}_{ij} depends on the relative geometrical configuration of elements i and j, and also temperature. It is assumed that there are no conductive heat losses to the surrounding environment.

2-4 CONVECTION

Convective cooling is provided by means of exhaust fans at the outlet end of the wind tunnel. The model is located in the "throat" of the tunnel, a region of constant cross section, 1" wide and 4 1/2" high.

The boundary-layer heat flux to the surface is computed using the Newton heat-transfer equation

$$q^{\text{conv}} = h (T_r - T_w) A \quad (5)$$

where

h is the heat transfer coefficient, BTU/in²/sec/°R

T_r is the recovery temperature, °R

T_w is the wall temperature, °R

A is the surface area, in²

Before discussing h, it is noted that the recovery temperature is computed with reference to the stagnation temperature, T_o (which in this case is the ambient, room temperature), and the temperature at the outer edge of the body layer, T_δ ,

$$T_r = T_\delta + r (T_o - T_\delta) \quad (6)$$

The recovery factor is dimensionless and taken to be 0.88, based on the discussion in Reference 1.

The free-stream temperature, T_δ , can also be computed from T_o , using perfect gas relations, where a_t represents the total speed of sound in the room.

$$a_t = \sqrt{1.4 (1716) T_o} = \sqrt{2402.4 T_o} \quad (7)$$

$$T_\delta = T_o \left\{ 1 - 0.2 (V/a_t)^2 \right\} \quad (8)$$

V is the free-stream velocity in ft/sec. The free-stream pressure, which will be needed later, can also be computed assuming a perfect gas,

$$P_\delta = P_o \left\{ 1 - 0.2 (V/a_t)^2 \right\}^{3.5} \quad (9)$$

where P_o is the ambient (room) pressure.

This, then, leaves the heat transfer coefficient, h , to be determined. This parameter is a function of many factors, but depends primarily on the type of flow in the region of interest, namely laminar vs. turbulent. This, in turn, depends on the tunnel geometry, roughness of the surface, any leaks in the tunnel, etc. Ideally, one probably would predict laminar flow in the test facility setup, but the discussion in Reference 4 indicates the flow is actually fully developed turbulent. Due to the factors mentioned above, this is plausible and accepted.

Based on turbulent flow, then, the Colburn equation (Reference 5) is used to take into account dependence of h on wall temperature and air-stream velocity (two velocities are possible, depending on whether one or two blowers is used). This requires either knowledge of the effective origin of the turbulent flow, or some empirical information regarding the value of h for a particular case. Reference 4 provided the latter, where an average value of 1.67×10^{-4} BTU/sec/in²/°R was determined for an air flow of 793 ft/sec. The effective origin (x_{eff}) was then back figured to be 1.15 inches.

The Colburn equation and other necessary relations are presented below. The user is referred to Reference 1 for a more complete discussion.

$$h = \frac{0.026 \cdot k \cdot R_{e\text{eff}}^{.8}}{x_{\text{eff}}} \quad (10)$$

where

$$k = \left[1.0 - 8.89 (10^{-6}) T' + 3.58 (10^{-8}) T'^{-2} \right] \cdot \left[2 (10^{-8}) T'^{-1.5} / (T' + 201.6) \right] \quad (11)$$

$$R_{e\text{eff}} = \frac{R_e \cdot V \cdot x_{\text{eff}}}{12} \quad (12)$$

$$R_e = 3.7 (10^6) P_\delta (T' + 198.6) / (T')^{2.5} \quad (13)$$

$$T' = 0.5 (T_W + T_\delta) + 0.22 (T_r - T_\delta) \quad (14)$$

The program also provides the user with the optional capability of specifying the heat transfer coefficient, h , which allows him more direct control over that parameter.

2-5 TEMPERATURE CHANGE

The computer program combines the results of all the heat transfer mechanisms at each instant of time for each element. The temperature change can then be calculated for the finite time interval according to

$$\Delta T = \frac{\Delta t}{\rho A c_p} q \quad (15)$$

where

ΔT is the temperature rise, °R

Δt is the time interval, sec

ρ is the element density, lbs/in³

c_p is the specific heat of the material, BTU/lb-°R

A is the cross-sectional area (volume) of the element, in²

q is the net heat flow into the element, BTU/sec.

SECTION 3

COMPUTER PROGRAM DESCRIPTION

3-1 PROGRAM OPERATION

The TRAP-ML program was written in FORTRAN IV and developed on the Control Data Corporation (CDC) 6600 computer. The program requires card input, which is described in Section 3-2, and produces time history response output, the subject of Section 3-3. The program also has the capability to produce a plot file of temperatures versus time. This data file then serves as input to a second program, APLLOT, which produces paper plots. This second program requires no card input of its own.

To execute and also generate plots, then, requires the execution of both TRAP-ML and APLLOT back to back. The primary reason for separating the two codes was to enable the user to run with less than a 60,000₈ cell core requirement. This makes it possible to operate the code from remote terminal under current WPAFB restrictions. A typical job control setup is shown in Table 3.

The routines for both programs are listed in Table 4. The basic plot routines are included in the APLLOT package, rather than calling on system routines, so that the program is less dependent on one particular computer.

TRAP-ML requires approximately 53,000₈ to load and execute. Logical files TAPE1 and TAPE6 are used for input and output, respectively. TAPE5 is used internally to store the input data subsequent to reading the data into the computer. The plot information is written to file TAPE8 and is automatically rewound prior to and upon completion of job. More than one case can be run during a job execution, in which case the plot output for each case represents a separate file on TAPE8. Approximately 8 cp seconds are required to compile TRAP-ML using "OPT=1" and "R=2" options.

Table 3. Typical job control sequence.

Job Control Cards	Comments
JOB CARD.....	
MAP,PART	
ATTACH,A,TRAPML.....	} Attach and execute TRAP-ML binary file
A.	
ATTACH,B,APLOT.....	} Attach and execute APLOT <u>if</u> plots are desired.
B.	
ROUTE,PLOT,.....	Direct the plot file to remote terminal
EXIT.	
7/8/9	
{ input data for TRAPML	
6/7/8/9	

Table 4. TRAP-ML and APLOT routines.

TRAP-ML	APLOT
TRAPML	APLOT
BLOCK DATA	FIRST
CONVEC	KALINE
DTEMP	KALE
FLUX	KANUM
INT1Z	KASYM
RPLOT	KAVANS
RPRINT	KAXIS
SETUP	
SPECT	
TIN	
TSTEP	
XHEAT	

Program APLOT only requires the file TAPE8 for input, but does generate output on TAPE6. The program also utilizes file TAPE40 for internal purposes. If on-line plots are requested, then the file PLOTS represents the resultant plot file, ready to be directed to a plotter.

APLOT requires approximately 50,000₈ cells of memory to load and execute. Compilation requires only about four cp seconds.

Execution times will depend on the size of the model, the time interval required, and the duration of the response requested, but generally program APLOT will execute for only a small fraction of the time used by TRAP-ML. The example problem discussed in Section 3-4 required about 40 cp seconds for both programs.

3-2 PROGRAM INPUT

TRAP-ML looks for BATCH input on file INPUT. The input data are specified in groups, where each group begins on a separate card. More than one card may be required for a group, however. The variable type and format corresponding to each data group is given in parentheses in the input instruction and is always in fields of 12. For convenience, floating point numbers can be left justified in the field as long as the exponent is right justified. Also, zero values can be replaced by a blank field. Columns 73 through 80 are not used for data and can be used for card identification or other purposes.

All input parameters, where appropriate, should be compared with the maximum dimensions provided for in the program, as delineated in Table 5. This is very important since the program does not attempt to check the input for all such violations. In Table 5, the input parameters are delineated with an asterisk (*) preceding the variable name.

The specific instructions for input are contained in Table 6. The remainder of this section will attempt to amplify on those instructions and provide insight into the applicability and limitations of the program.

Table 5. Description of variables which determine program dimensions.

VARIABLE	LIMIT	DESCRIPTION
*NAT	10	Number of temperatures at which absorptivity is given.
NCONDC	122	Number of conduction connections between elements.
*NCPT	10	Number of temperatures at which specific heat is given.
*NCT	10	Number of temperatures at which conductivity is given.
NEL [#]	75	Total number of elements.
NFALL	8	Number of points at which the relative flux is specified versus time in the decay phase.
*NFLUX	15	Number of points describing thermal flux time history.
NLAY	5	Number of layers in segment.
NMAT	9	Number of different materials.
NRISE	8	Number of points at which the relative flux is specified versus time in the "rise" phase.
*NSEG	3	Number of segments.
*NTSPEC	3	Number of times at which spectral distribution of radiation source is specified.
NWL	14	Number of wavelength bands used to describe radiation source in calculations.
*NWLA	10	Number of wavelengths at which absorptivity is specified for new material.
*NWLI	50	Number of wavelengths used to describe spectral distribution.

NOTES:

- * - Variable is specified on input.
- # - Dimension also limited in subroutine SETUP by a fixed point number before statement number 23.

Table 6. TRAP-ML input.

Group 1: (40A2) (ID(I), I=1, 40)
 Title card - free field. (ID)

Group 2: (2I12) INOUT, IPLOT
 Program output-option code: (INOUT)
 0, do not print input data
 1, print input data
 Plot code (IPLOT)
 0, do not generate plots of temperature vs. time
 1, do generate plots of temperature vs. time

Group 3: (I12) NSEG
 Number of segments. ($1 \leq \text{NSEG} \leq 3$)

Group 4: (2F12.1) XSEG(IS), YSEG(IS)
 x-coordinate of end of segment, in (XSEG)
 y-coordinate of end of segment, in (YSEG)

Note - See Figure 3 for definition of coordinates.

Repeat Group 4 for IS=1, NSEG+1

Groups 5-7 provide data for each segment, IS, and are to be repeated for IS=1, NSEG

Group 5: (I12) NLAY(IS)
 Number of layers in segment ($1 \leq \text{NLAY}(\text{IS}) \leq 5$).

Groups 6-7 provide data for each layer, IL, and are to be repeated for IL=1, NLAY(IS)

Group 6: (F12.1) TLAY (IS,IL)
 Thickness of layer, in. [TLAY(IS,IL)]

Group 7: (2I12) KMAT (IS,IL), NELL(IS,IL)
 Code defining material of layer: (KMAT (IS,IL))
 1, Aluminum, 2024-T3
 2, Aluminum, 7075-T6
 3, Magnesium, AZ31B-H24
 4, Titanium, Ti-8Mn

Table 6. (Continued)

If materials other than those listed above are used, they should be assigned numbers which are consecutive beginning with 5. Material properties for the added materials are called for in Groups 8-21

Number of elements in layer. (NELL(IS,IL))

Groups 8-22 provide material properties data for each added material, NM, and should be repeated until all added materials have been completed. If there are no added materials, skip to Group 23.

Group 8: (I12) MCOR(NM)

Number of material to which new material partially corresponds. This number may be any number lower than the number of the new material for which data are now being supplied. Use of a corresponding material allows the analyst to skip reading in material properties which are the same as those of the corresponding material. If this number is read in as zero, all material properties for the new material must be supplied, starting with Group 10 and ending with Group 22. (MCOR(NM))

If MCOR = 0, skip to Group 10.

Group 9: (I12) KCH

Code indicating material property to be changed:

- 1, absorptivity
- 2, conductivity
- 4, specific heat
- 5, melting temperature
- 6, heat of fusion
- 11, density

Under this mode of input, a change card is read in for each property to be changed, followed by data for the specified material property. Group 22 terminates this sequence. (KCH)

Groups 10-13 provide absorptivity data for the new material, if required.
(MCOR = 0 or KCH = 1)

Group 10: (I12) NAT(NM)

Number of temperatures at which absorptivity is given
NAT(NM))

Table 6. (Continued)

Groups 11-13 provide data for one temperature, NT, and are to be repeated for NT = 1, NAT(NM).

Group 11: (2F12.1) TALF(NT,NM), EMI(NT,NM)

Temperature at which absorptivity is given, °R
(TALF(NT,NM))

Emissivity at specified temperature, dimensionless
(EMI(NT,NM)) (Omit if integration of absorptivity by
program is desired in order to obtain emissivity)

Group 12: (I12) NWLA

Number of wavelengths at which absorptivity is specified
(NWLA)

Group 13: (2F12.1) WL(NW,NT), ALF(NW,NT)

Wavelength at which absorptivity is given, microns*
(WL(NW,NT))

Absorptivity, dimensionless (ALF(NW,NT))

Repeat Group 13 for NW = 1, NWLA.

Groups 14-16 provide conductivity data for the new material, if required (MCOR = 0 or KCH = 2)

Group 14: (I12) KLD(NM)

Code of conductivities: (KLD(KM))

- 1, lengthwise and depthwise conductivities are the same
- 2, lengthwise and depthwise conductivities are different

Group 15: (I12) NCT(NM)

Number of temperatures at which conductivity is given
(NCT(NM))

Group 16: (3F12.1) TCOND(NT,NM), CONDL(NT,NM), CONDD(NT,NM)

Temperature at which conductivity is given, °R (TCOND(NT,NM))

Lengthwise conductivity of material, BTU/in-sec-°R
(CONDL(NT,NM))

*If emissivity has been specified in Group 11, only wavelengths from 0.2 to 5.5 microns must be covered. If emissivity is to be found by integration of absorptivity, wavelengths from 0.2 to about 250 microns must be covered.

Table 6. (Continued)

Depthwise conductivity of material (Omit if KLD(NM) = 1),
BTU/in-sec-°R (CONDD(NT,NM))

Repeat Group 16 for NT = 1, NCT(NM)

Groups 17-18 provide specific heat data for the new material, if
required (MCOR = 0 or KCH = 4)

Group 17: (I12) NCPT(NM)

Number of temperatures at which specific heat is given
(NCPT(NM))

Group 18: (2F12.1) TCP(NT,NM), CPM(NT,NM)

Temperature at which specific heat is given, °R
(TCP(NT,NM))

Specific heat of material, BTU/lb-°R(CPM(NT,NM))

Repeat Group 18 for NT = 1, NCPT(NM)

Group 19 provides melting temperature data for the new material if
required (MCOR = 0 or KCH = 5)

Group 19: (F12.1)

Melting temperature, °R (TMELT(NM))

If the melting temperature is changed, usually all of the material properties will have to be changed. The reason for this is as follows: the melting temperature for each material for which data are provided in the program is the same as the maximum temperature for the material property data. Suppose that the melting temperature is increased. Unless the other material properties are also changed, the melting temperature will be above the maximum temperature at which material properties are available. The program will then produce an error stop.

Group 20 provides heat of fusion data for the new material, if required
(MCOR = 0 or KCH = 6)

Group 20: (F12.1) HOF(NM)

Heat of fusion of material, BTU/lb (HOF(NM))

Table 6. (Continued)

Group 21 provides density data for the new material, if required
(MCOR = 0 or KCH = 11)

Group 21: (F12.1) RHOM(NM)

Density of material, lbs/in^3 (RHOM(NM))

Group 22: Blank card. Is required for each new material NM, either at the completion of Groups 10-21 for MCOR(NM) = 0, or at the completion of the sets of data beginning with Group 9 for NCOR(NM) 70.

Group 23: (4F12.1) VEL, PO, TAM, HCONO

Velocity of free-stream flow, ft/sec (VEL)

Ambient (room) pressure, psi (PO)

Ambient (room) temperature, degrees F (TAM)

Convective heat transfer coefficient, $\text{BTU/in}^2\text{-sec-}^\circ\text{R}$ (HCONO)

Note - Program computes coefficients for each segment
if HCONO=0.0.

Group 24: (I12) NOPT

Code designating thermal flux time history (NOPT)

1, 6000 watt tungsten source (Figure 2)

2, Step load.

3, Arbitrary point-by-point description.

Group 25: (F12.1) QDMAX

Maximum flux achieved, $\text{BTU/in}^2\text{-sec}$ (QDMAX)

If NOPT=3, skip Group 26

Group 26: (F12.1) TCUT

Time at which thermal source is cut off, sec. (TCUT)

If NOPT<3, skip Groups 27-28.

Group 27: (I12) NFLUX

Number of points describing thermal flux time history.

First point must include time=0. ($2 \leq \text{NFLUX} \leq 15$)

Table 6. (Continued)

Group 28: (2F12.1) TFLUX(I), FLUXT(I)

Time, sec (TFLUX(I))

Relative flux corresponding to TFLUX, BTU/in²/sec.
Magnitude of flux is adjusted to QDMAX. (FLUXT(I))

Repeat Group 28 for I = 1, NFLUX

Group 29: (2I12) NWLI, NTSPEC

Code designating spectral distribution of flux (NWLI)
0, 6000 watt tungsten characteristics for all
times (Figure 1).

>0, number of wavelengths used to describe distribution
(2<NWLI<50).

Number of times at which spectral distribution is specified.
Assumed to be 1 if NWLI=0. (1<NTSPEC<3).

If NWLI=0 or NTSPEC=1, skip Group 30.

Group 30: (6F12.1) (TSPEC(I), I=1, NTSPEC)

Times at which spectral distribution is specified.
First time must be 0.0. (TPSEC)

If NWLI=0, skip Group 31.

Group 31: (2F12.1) (WAVE(J,I), QR(J,I))

Wavelength at which flux is specified. First wavelength
must be 0.2 microns; the last must be 5.5. (WAVE)

Relative flux corresponding to WAVE, BTU/in²/sec. Only
relative magnitudes are important. (QR)

Repeat Group 31 for J=1, NWLI, then repeat that sequence for I=1, NTSPEC.

Group 32: (3F12.1) DELTIM, TSTOP, PRINT

Integration time interval (if read in as 0.0, program will
determine time interval required for stability), sec
(DELTIM)

Time at which computations are to stop, sec (TSTOP)

Number of time intervals between printouts (0.0 will
give no printout), dimensionless (PRINT)

Table 6. (Concluded)

To execute another problem, repeat Groups 1-32.

Group 33: (A3) IEND

To terminate job, place "END" in columns 1-3. (IEND)

TRAP-ML has the capability to assemble a two-dimensional heat transfer model of a particular material specimen. This implies that the specimen and the thermal loads are uniform in one coordinate direction - defined here to be the z direction. The x-coordinate direction is defined as in the direction of the air flow, and y is defined to be from the unexposed surface through the cross section toward the exposed surface (see Figure 3).

The model consists of segments, layers and elements, where elements are the basic "building blocks" and must be thermally "thin"; i.e., the temperature gradient is accurate only to the extent that enough elements are included in the model. Geometrically, the elements are horizontal slices of the cross section of constant thickness. Layers consist of a set of elements, all of which exhibit identical material properties. The model can contain up to five layers in the cross section per segment. The segment is defined as a depthwise set of layers representing a rectangular portion of the cross section. Elements, layers, and segments all have constant thickness. If only one segment is included in the model, the program is, by definition, reduced to a one-dimensional analysis. It is expected that only one segment will be used to model the test specimens.

In the general model indicated in Figure 3, segment 1 has only one layer and five elements. Segment 2 consists of two layers, each having two elements.

Now for the input instructions in Table 6. Group 1 represents a free-field case descriptor which appears as a title in the output. The first 40 columns will also appear on plots if requested. It should be pointed out that data groups 1 through 32 are repeated for as many cases as the user desires.

Group 2 defines the output options. By selecting INOUT=1, the output is reduced somewhat and the user still gets a raw data listing at the beginning of the output. The plots which are generated if IPLOT=1 give temperature at both surfaces and at each layer boundary versus time.

Group 4 defines the location and size of each segment by specifying the lower, left-hand and right-hand coordinates of each segment as defined in Figure 3. The coordinate origin can be located anywhere - one might just as well start with (0., 0.) for the first segment as not. Groups 5 and 6 define the layers.

Group 7 introduces the material properties. If materials or material properties other than those associated with the four materials indicated are desired, the user can accomplish this by calling for one or more additional materials. The first new material is number 5, the second number 6, etc. The properties corresponding to the four built-in metal alloys are indicated in Table 7 and Figures 4 and 5. In this regard, it will be noted that the selection of these four materials was arbitrary and the user is referred to Reference 1 for a complete discussion of property specification. It should be noted, for example, that the absorptivities specified for materials 1-4 are all 0.5 for an unpainted condition. If the user wants to specify a different absorptivity, he would need to call for a new material and proceed with the input.

Also in Group 7 the user must decide on the number of elements per layer. The program automatically divides each layer into elements of equal thickness, depending on the number (NELL) specified. In general, thicker layers with lower conductivities will require more elements. One guideline, other than checking the resulting output to see that the temperature gradient is not too large, is to check the Δt calculation (see the discussion of Δt in Group 32). Here it makes some sense to end up with a Δt requirement more or less the same for all elements. At least two elements per layer are recommended so that a good estimate of temperature can be made at the boundary between layers and at the surface.

Group 32 requires the selection of a Δt to be used in the numerical integrations. This number must be small enough to give a stable solution. If $\Delta t=0.0$ is specified, the program will compute a Δt associated with each element and then select the smallest of these.

Table 7. Material properties in TRAP-ML.

Material Property	Aluminum 2024-T3	Aluminum 7075-T6	Magnesium AZ31B-H24	Titanium Ti-8Mn
Absorptivity	0.5	0.5	0.5	0.5
Emissivity	0.5	0.5	0.5	0.5
Heat of Fusion (BTU/lb)	170.	170.	158.	188.
Density (lb/in ³)	0.100	0.101	0.064	0.171
Melt Temp. (°R)	1680.	1680.	1662.	3495.

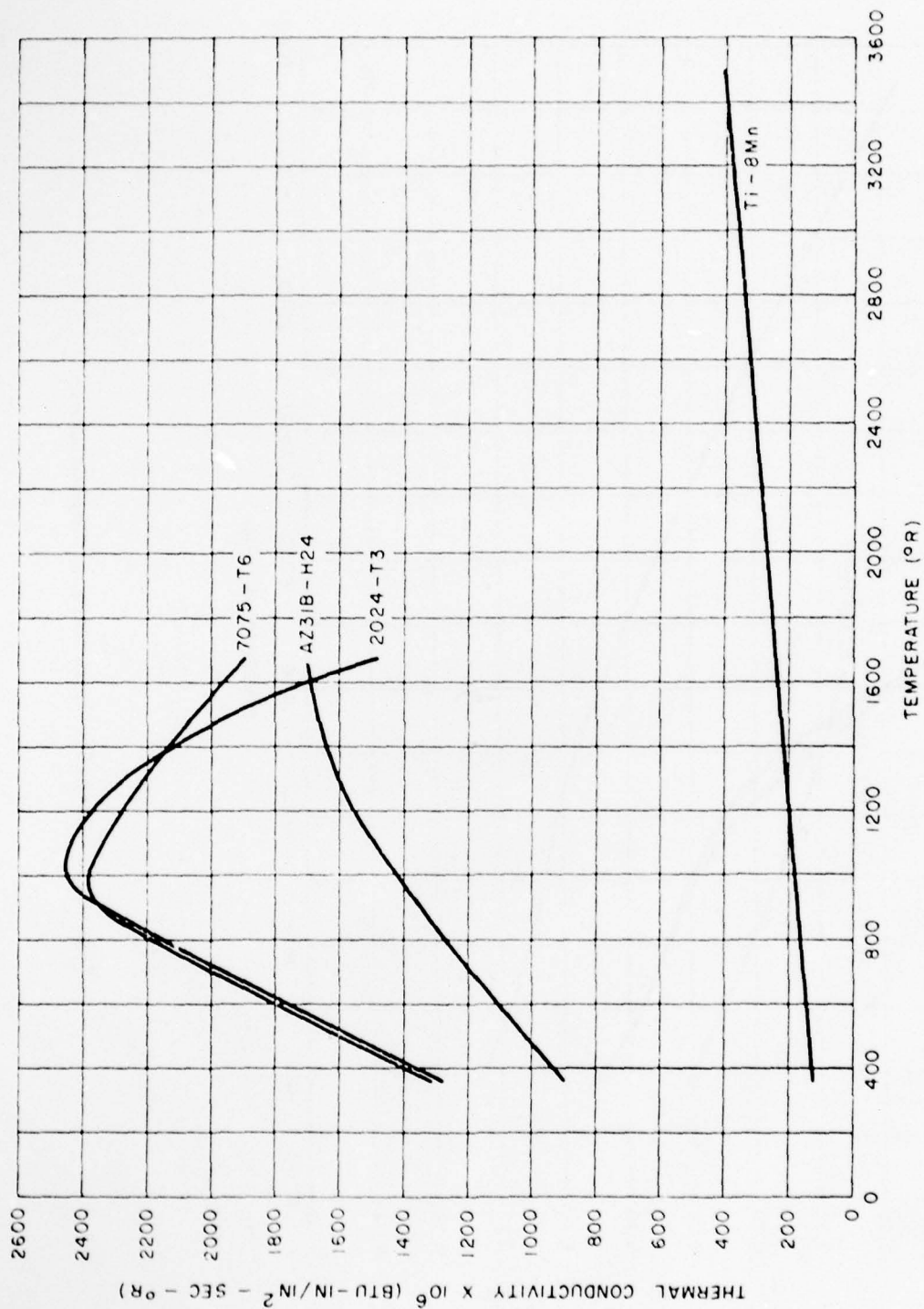


Figure 4. Thermal conductivity versus temperature.

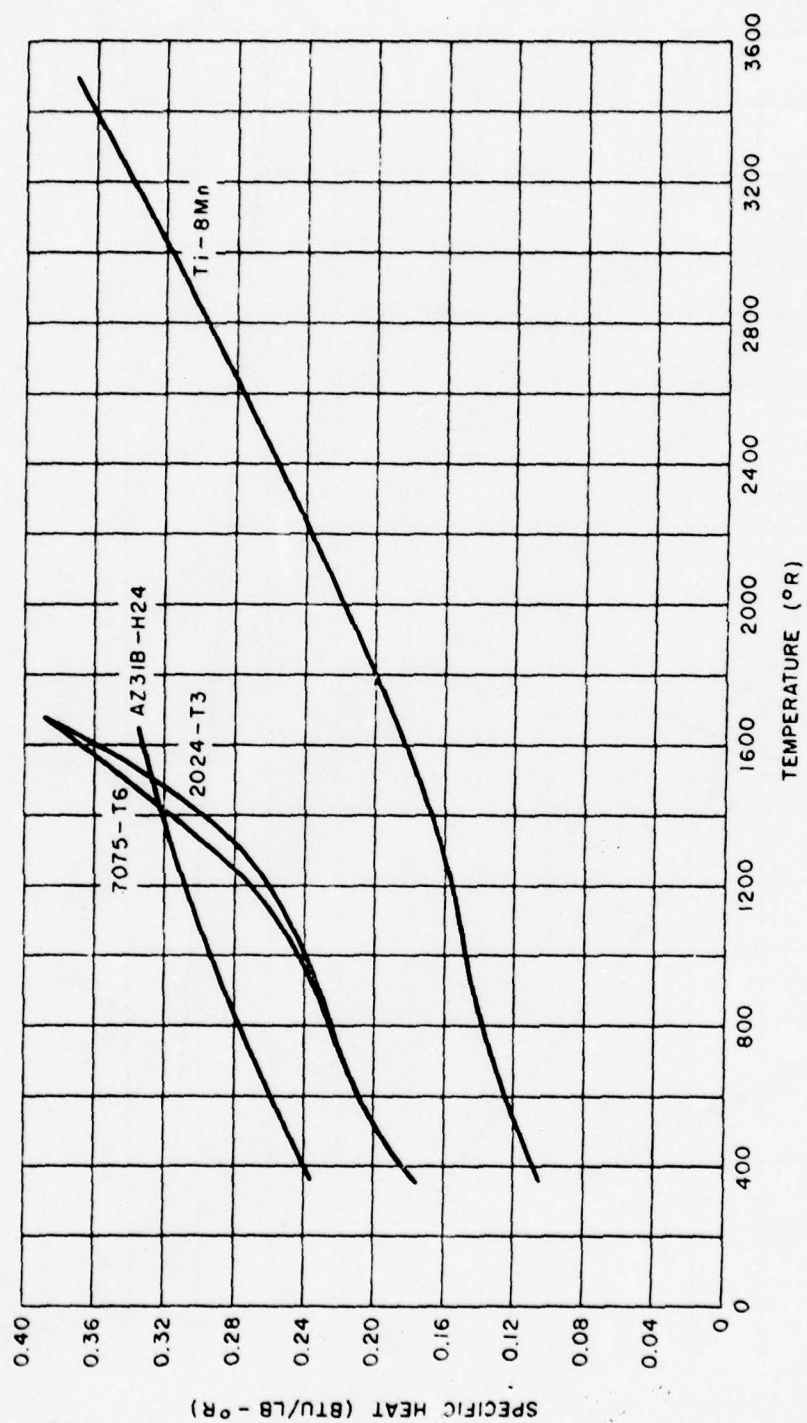


Figure 5. Specific heat versus temperature.

The user is referred to Reference 1 for a complete discussion of the formulation, but the resultant formula used is the following:

$$\Delta t_i = \frac{0.8 \rho_i A_i C_{p_i}}{\sum_j \bar{K}_{ij}} \quad (16)$$

where the summation on j extends over all elements which touch element i. These numbers are then printed out for the user's information.

Groups 8 through 22 permit the user to specify new material properties. Note that if the new material (called for in Group 7) does not have totally different properties from one of the four basic material options, then only those properties that change need be specified. For example, if one were to specify the absorptivities for 7075-T6 Al, then MCOR (Group 8) would be 2 and KCH (Group 9) would be 1. The absorptivities would be specified in Groups 10-13.

With regard to melt temperature (Group 19), the program treats the melting of an element by eliminating it from the solution. The next element then becomes exposed to the radiation source, with the implicit assumption that the melting always proceeds from the exposed surface toward the opposite surface, element by element. The output reflects this event by setting the temperature of a melted layer to 0.0 in the printed output, and a small vertical arrow indicates the event on the plotted output.

3-3 PROGRAM OUTPUT

TRAP-ML output consists of three logically distinct sections. The first output is a card image listing of the entire input deck, exactly as it is read in, except that the cards are numbered. The second section describes the input data (if INOUT=1) in terms of its meaning in the program.

The third section is the time-history response output. At each printout time the following values are printed: time, incident flux, incident fluence, and a table of (1) temperatures and (2) a

breakdown of heat flow factors for each element. Figure 6 represents typical time-history output for three consecutive printouts.

If plots are requested (IPLOT=1), the program also stores between 100 and 200 sets of temperatures (evenly spaced in time) for both surfaces of the model and for each layer division for segment 1, or for segment 3 if there are 3 segments in the model. Linear interpolation/extrapolation is used to estimate these temperatures based on the available values at the center of each element. Each curve plotted is identified by a unique symbol. These symbols are defined in Table 8. The final two lines of printed output indicate the number of data points per curve and the number of curves written to disk file.

Error messages are largely self-explanatory, although the linear interpolation routine (INT1Z) only indicates a code number indicating the source of a problem if a variable exceeds the limit of the table. Table 9 identifies the routine corresponding to the error code.

Program APLOT generates a list of the data plotted and indicates the completion of each graph as it is actually generated.

```

TIME, SEC = .158437E+01
INCIDENT FLUX, BTU/IN**2/SEC = .818400E+00
INCIDENT FLUENCE, BTU/IN**2 = .119360E+01

SEGMENT ELEMENT TEMPERATURE
NUMBER NUMBER DEGREES R
1 1 .7407E+03
1 2 .7145E+03
1 3 .8066E+03
1 4 .8593E+03
1 5 .8304E+03
1 6 .8157E+03
1 7 .8153E+03

H E A T F L O W , B T U / S E C
ABSORPTION RADIATION CONVECTION CONDUCTION
.7408E+00 -.8184E-02 -.2827E+00
.7476E-02
.7476E-02
.7681E-02
.7681E-02
.8079E-02
.8079E-02
.8079E-02
.2342E+00
.2342E+00
.2344E+00

NET
.7266E-02
.7476E-02
.7681E-02
.7681E-02
.8079E-02
.8079E-02
.8079E-02
.2342E+00
.2344E+00

TIME, SEC = .198040E+01
INCIDENT FLUX, BTU/IN**2/SEC = .819855E+00
INCIDENT FLUENCE, BTU/IN**2 = .151703E+01

SEGMENT ELEMENT TEMPERATURE
NUMBER NUMBER DEGREES R
1 1 .1005E+04
1 2 .9763E+03
1 3 .9480E+03
1 4 .9201E+03
1 5 .8927E+03
1 6 .8744E+03
1 7 .8744E+03

H E A T F L O W , B T U / S E C
ABSORPTION RADIATION CONVECTION CONDUCTION
.7436E+00 -.1030E-01 -.2909E+00
.6951E-02
.6951E-02
.7135E-02
.7135E-02
.7316E-02
.7492E-02
.7492E-02
.2264E+00
.2264E+00
.2304E+00

NET
.6764E-02
.6951E-02
.7135E-02
.7135E-02
.7316E-02
.7492E-02
.7492E-02
.2264E+00
.2304E+00

TIME, SEC = .237655E+01
INCIDENT FLUX, BTU/IN**2/SEC = .101594E+00
INCIDENT FLUENCE, BTU/IN**2 = .166297E+01

SEGMENT ELEMENT TEMPERATURE
NUMBER NUMBER DEGREES R
1 1 .4658E+03
1 2 .4734E+03
1 3 .4803E+03
1 4 .4864E+03
1 5 .4932E+03
1 6 .4984E+03
1 7 .4984E+03

H E A T F L O W , B T U / S E C
ABSORPTION RADIATION CONVECTION CONDUCTION
.9434E-01 -.5694E-02 -.2191E+00
.1207E+00
.5115E-02
.4299E-02
.3365E-02
.2361E-02
.5515E-01
.5515E-01
.5469E-01

NET
-.5781E-02
-.5115E-02
-.4299E-02
-.3365E-02
-.2361E-02
-.5515E-01
-.5515E-01
-.5469E-01

```

Figure 6. Time-history output for sample problem.

Table 8. Plot symbols used to identify plot curves.

Curve Number (front face first- back face last)	Plot Symbol
1 (front face)	□
2	△
3	×
4	◇
5	+
6	○

Note - A vertical arrow (↑) indicates the time at which the element melted.

Table 9. Error codes from INT1Z.

Error Code	Subroutine which called INT1Z	Location Code #
1	SPECT	300^{-2}
2	FLUX	300
3	SPECT	700^{-1}
4	SETUP	228^{-3}
5	TSTEP	12
6	TSTEP	13
7	DTEMP	7
8	DTEMP	10
9	XHEAT	62
10	XHEAT	83
11	DTEMP	25
12	SETUP	117^{-1}
13	TSTEP	12^{-4}
14	XHEAT	42

#The location code is read as follows: a code $S^{\pm n}$ is interpreted as FORTRAN statement number S plus or minus n cards.

SECTION 4

SAMPLE PROBLEM AND COMPARISON WITH EXPERIMENT

The sample problem described in this section is designed to (a) provide the user with a check case for exercising the code, (b) provide an example of input data preparation, and (c) make a limited comparison with experimental results. The example case selected is based on an experiment performed at the DNA test facility and documented in Reference 4. A four inch sample of 2024-T3 aluminum substrate .032 inches thick was coated with .003 inch of white polyurethane and subjected to a 793 ft/sec wind tunnel flow and a two second pulse from the quartz lamp bank.

The input data for TRAP-ML is indicated in Figure 7. Although the data corresponds to the test data as closely as possible, the documentation was not totally complete. Figure 8 represents the plot generated by APLLOT of three temperature-time histories: (1) the exposed surface, which obtained a peak temperature of 1019°R, (2) the division between coating and substrate, and (3) the unexposed surface (backface), which obtained a peak temperature of 900°R. The last two curves fall virtually one on top of the other due to the relatively high conductivity of the aluminum. Remember, the plot symbol codes are tabulated in Table 8. The printed output in the vicinity of where peak front-face temperatures were computed can be found in Figure 6.

Figure 9 compares the backface temperature with that presented in Figure 26 of Reference 4. Also shown are the results of the ASTHMA code from Reference 4. In general, TRAP-ML seems to predict the temperature response well - its only shortcoming is a failure in this case to predict as large a peak temperature. This minor deviance could easily be explained by any number of uncertainties in the data, e.g., the exact shape of the radiation pulse time history.

WHITE POLYURETHANE OVER .032 IN. 2024-T3 SAMPLE PROBLEM.

	1	1	
	1		
0.	0.		
4.	0.		
	2		
.003			
	5	5	
.032			
	1	2	
	0		
	2		
460.	.758		
	2		
.2	.242		
5.5	.242		
4000.	.758		
	2		
.2	.242		
5.5	.242		
	1		
	2		
0.	2.54	E-6	
4000.	2.54	E-6	
	2		
0.	.41		
4000.	.41		
1200.			
50.			
.0502			
793.	14.3	75.	0.
	1		
.82			
2.			
	0	0	
0.	8.	500.	
END			

Figure 7. Sample problem input data.

WHITE POLYURETHANE OVER .032 IN. 2024-T3
PEAK FLUX, BTU/IN²/SEC = .820
PULSE DURATION, SEC = 2.000

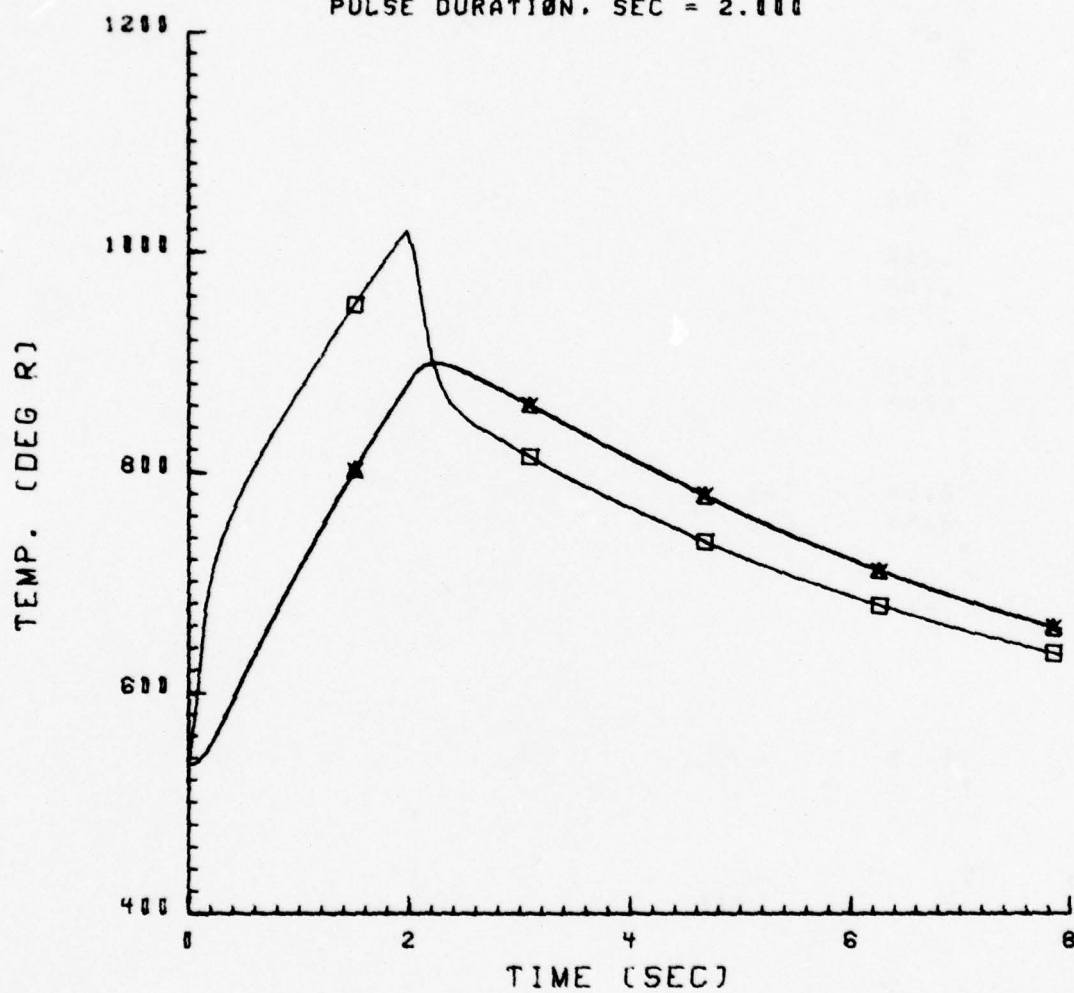


Figure 8. Computer-generated sample plot.

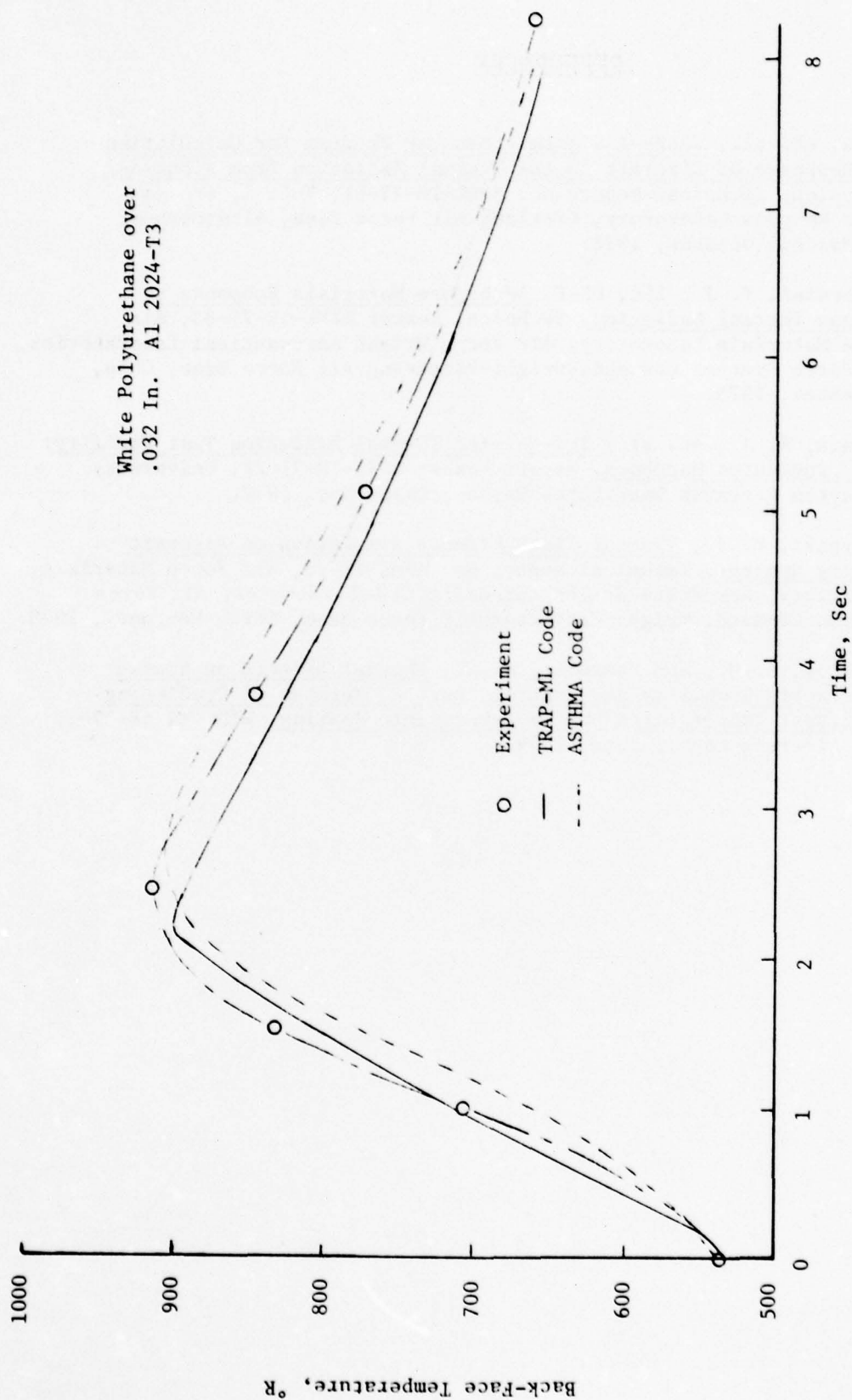


Figure 9. Comparison of TRAP-ML predictions to experiment.

REFERENCES

1. Hobbs, et. al., TRAP--A Digital Computer Program for Calculating the Response of Aircraft to the Thermal Radiation From A Nuclear Explosion, Technical Report No. AFWL-TR-71-61, Vol. 1, the Air Force Weapons Laboratory, Kirtland Air Force Base, Albuquerque, New Mexico, October, 1972.
2. Vandersaar, F. J., 1Lt, USAF, Aerospace Materials Response to Intense Thermal Radiation, Technical Report AFML-TR-75-83, Air Force Materials Laboratory, Air Force Wright Aeronautical Laboratories, Air Force Systems Command, Wright-Patterson Air Force Base, Ohio, September, 1975.
3. Servais, R. A., et. al., Tri-Service Thermal Radiation Test Facility: Test Procedures Handbook, Report Number UDRI-TR-77-28, University of Dayton Research Institute, Dayton, Ohio, May, 1977.
4. Moscynski, M. J., Thermal Flash Effects Simulation on Aircraft Coating Systems, Technical Report No. AFML-TR-78, Air Force Materials Laboratory, Air Force Wright Aeronautical Laboratories, Air Force Systems Command, Wright-Patterson Air Force Base, Ohio, February, 1978.
5. Blessing, A. H., and Hanawalt, A. J., Thermal Effects on Static Aeroelastic Stability and Control, Part I, Methods of Predicting Structural Temperatures Due to Aerodynamic Heating, WADC TR 58-378, Bell Aircraft Corp., June, 1959.

APPENDIX A
PROGRAM LISTING OF
TRAP-ML

```

PROGRAM TRAPML (INPUT,OUTPUT,TAPE1=INPUT,TAPE6=OUTPUT,
1 TAPE5=129,TAPE8=514)

```

TRAP-ML IS A SPECIAL VERSION OF TRAP DEVELOPED FOR THE AIR FORCE MATERIALS LABORATORY, WPAFB, BY KAMAN AVIDYNE, BURLINGTON, MA. VERSION 1.0, SEPT., 1978. TEMPERATURE VS. TIME PLOT INFORMATION IS WRITTEN TO TAPE8. PROGRAM APLOT IS REQUIRED TO GENERATE THE PLOTS.

```

COMMON A(75),ALF(10,10),B(122),CKD(75),CKDM(9),CKL(75),CKLM(9),
1 CONDDAG(122),CONDD(10,9),CONDI(122),CONDJ(122),CP(75),CPMAT(9),
2 DELTIM,DTIM(75),DTIME,DTMIN,EL(75),FLUXT(15),ICONO,IJ(20),
3 IEL(3),IELAY(5,3),IEND,IMAT(75),IMELT(75),INOUT,IPLT,ISEG(75),
4 ISTOP,JEL(3),KCH,KCONDI(122),KCONDJ(122),KCPI(122),KCPJ(122),
5 KMAT(3,5),KSJR,LAYN(75),MCON,NCONDC,NEL,NELL(3,5),NELS(3),
6 NFLUX,NLAY(3),NMAT,NOPT,NPRINT,NSEG,NTSPEC,NW,A,NWLI,PO,
7 PRINT,Q(75),QAB(3),QCOND(75),QCONV(3),QDMAX,QFLU,QR(50,3),
8 QRAD,QRER(2,3),QPR(14),SEGL,T(75),TAM,TCUT,TEF(3),TEMP(75),
COMMON TFLUX(15),TFS,TIME,TLAY(3,5),TP(75),TR,TSPEC(3),TSTOP,
1 VEL,WAVE(50,3),WAVEL(14),WL(10,10),WT(75),X(75),XCON(75),
2 XSEF(4),XTR,Y(75),YSEG(4)
COMMON /BLOCK/ ALFAT(10,14,9),CONDL(10,9),CPM(10,9),EMI(10,9),
1 QDF(9),KLD(9),NAT(9),NCPT(9),NCT(9),NMATT,NWL,PI,RHOM(9),
2 TALF(10,9),TCOND(10,9),TCP(10,9),TMELT(9),WAVEB(15)

CALL TIN

5 CALL SETUP
IF (ISTOP.EQ.1) GO TO 100
IF (DELTIM.EQ.0.0) CALL TSTEP
CALL XHEAT(0)
DO 10 IE=1,NEL
TEMP(IE) = TAM
IMELT(IE) = 0
10 Q(IE) = 0.0
TIME = 0.
NPRINT = 0
QFLU = 0.

CALL XHEAT(2)
CALL DTEMP(0)
IF (IPLT.GT.0) CALL RPLT(0)

70 IF (IPLT.GT.0) CALL RPLT(1)
IF (NPRINT.GT.0) GO TO 75
IF (PRINT.EQ.0.0) GO TO 77
CALL PPRINT
NPRINT=PRINT-0.5
GO TO 77
75 NPRINT=NPRINT-1
77 TIME=TIME+DELTIM
IF (TIME.GT.TSTOP) GO TO 90
CALL XHEAT(2)
CALL DTEMP(1)
QFLU = QFLU + QRAD*DELTIM
GO TO 70

90 IF (IPLT.GT.0) CALL RPLT(2)
GO TO 5

100 ENDFILE 8
REWIND 8
120 STOP
END

```

TRAPML

BLOCK DATA

COMMON /BLOCK/ ALFAT(10,14,9),COND(10,9),CPM(10,9),EMI(10,9),
1 4DF(9),KLD(9),NAT(9),NCPT(9),NCT(9),NMATT,NWL,PI,RHOM(9),
2 TALE(10,9),TCOND(10,9),TCP(10,9),TMELT(9),WAVEB(15)

ABSORPTIVITY.

DATA NAT/4*2/,ALFAT/560*0.5/,TALF(1,1)/300.0/,
1 TALF(2,1)/1680.0/,TALF(1,2)/300.0/,TALF(2,2)/1680.0/,
2 TALF(1,3)/300.0/,TALF(2,3)/1662.0/,TALF(1,4)/300.0/,
3 TALF(2,4)/3495.0/

SPECIFIC HEAT, BTU/LB-DEG R.

DATA NCPT/9,9,8,9/

MATERIAL 1, 2024-T3.
DATA CPM/0.175,0.191,0.214,0.229,0.244,
10.259,0.287,0.313,0.337,0.357,

MATERIAL 2, 7075-T6.
20.174,0.190,0.213,0.230,0.250,
30.231,0.307,0.332,0.359,0.389,

MATERIAL 3, A231B.
40.237,0.246,0.264,0.281,0.298,
50.312,0.324,0.336,0.336,0.336,

MATERIAL 4, TI-8MN.
60.104,0.1275,0.146,0.153,0.158,
70.154,0.204,0.251,0.370,0.370/

MATERIAL 1, 2024-T3.
DATA TCP/360.0,460.0,660.0,860.0,1060.0,
11260.0,1360.0,1460.0,1680.0,1680.0,

MATERIAL 2, 7075-T6.
2360.00,460.00,660.00,860.00,1060.0,
11260.0,1360.0,1460.0,1680.0,1680.0,

MATERIAL 3, A231B.
4360.00,460.00,660.00,860.00,1060.0,
51260.0,1460.0,1662.0,1662.0,1662.0,

MATERIAL 4, TI-8MN.
6360.00,660.00,960.00,1160.0,1260.0,
71350.0,1860.0,2760.0,3495.0,3495.0/

THERMAL CONDUCTIVITY, BTU/IN-SEC-DEG R.

DATA NCT/10,10,3,2/

MATERIAL 1, 2024-T3.
DATA CONDL/0.001283,0.001875,0.002267,0.002421,0.002450,
10.002411,0.002323,0.002189,0.002020,0.001485,

MATERIAL 2, 7075-T6.
20.001312,0.001890,0.002089,0.002284,0.002377,
30.002359,0.002240,0.002091,0.002000,0.001895,

MATERIAL 3, A231B.
40.000901,0.000930,0.001161,0.001325,0.001470,
50.001584,0.001659,0.001680,0.001700,0.001700,

MATERIAL 4, TI-8MN.
60.000127,9*0.000405/

MATERIAL 1, 2024-T3.
DATA TCOND/360.0,660.0,860.0,960.0,1060.0,
11150.0,1260.0,1360.0,1460.0,1680.0,

MATERIAL 2, 7075-T6.
2360.00,660.00,760.00,860.00,960.00,
31060.0,1260.0,1460.0,1560.0,1680.0,

MATERIAL 3, A231B.
4360.00,460.00,660.00,860.00,1060.0,
51260.0,1460.0,1560.0,1662.0,1662.0,

MATERIAL 4, TI-8MN.
6360.0,9*3495.0/

EMISSIVITY.

DATA EMI/40*0.5/

BLOCKD

C HEAT OF FUSION.

DATA HOF/170.0,170.0,150.0,180.0/

C DENSITY.

DATA PHOM/0.100,0.101,0.064,0.171/

C MELT TEMPERATURES FOR MATERIAL GIVEN, DEGREES R.

DATA TMELT/1680.0,1680.0,1662.0,3495.0/

DATA KLD/4*1/, HMA77/4/

DATA NWL/14/

DATA WAVEB/0.2, 0.375, 0.625, 0.875, 1.125, 1.375, 1.625, 1.875,
1 2.25, 2.75, 3.25, 3.75, 4.25, 4.75, 5.25/

DATA PI/3.1415926535898/

END

BLOCKD

```

SUBROUTINE CONVEC (IFIRST,PO,TAM,VEL,TEMP,EL,XCON,HCONQ,TE,
1 TR,QCON)
THIS ROUTINE CALCULATES THE CONVECTION LOSSES TO TO AIR-STREAM
COOLING.
DIMENSION TEMP(75), EL(75),XCON(75)
IF (IFIRST.GT.0) GO TO 40
TAM = TAM + 453.62
VC = AMAX1(VEL,30.)
TS = TAM
AT = SQRT(2402.4*TS)
TFS = TS*(1.0-0.2*(VC/AT)**2)
PAM = PO*(1.0 - (.2*(VC/AT)**2)**3.5)
TR = TFS + .38*(TS - TFS)
XEFF = 1.15
WRITE (6,10) TR
10 FORMAT ('2HRECOVERY TEMPERATURE, DEG. R = F15.6)
RETURN
CALCULATE CONVECTIVE HEAT FLOW.
40 HCONQ = HCONQ
IF (HCONQ.GT.0.0) GO TO 60
TPRIM = .5*(TEMP(IE) + TR) + .22*(TR-TFS)
PHI = 3.7F6*PAM*(TPRIM+198.6)/TPRIM**2.5
AK=(1.0-8.83E-6*TPRIM+7.58E-8*TPRIM**2)*2.0E-8*TPRIM**1.5/
1 (TPRIM+281.6)
RE=PHI*VC*XEFF/12.0
HCONQ=0.026*AK*RE**0.5/XEFF
60 QCON = HCONQ*(TR-TEMP(IE))*EL(IE)
RETURN
END

```

CONVEC

SUBROUTINE DTEMP(KC)

```

C      DTEMP CALCULATES HEAT FLOW BY CONDUCTION AND INTERNAL RADIATION
C      AND UPDATES TEMPERATURES FOR EACH ELEMENT.

COMMON A(75), ALF(10,10), B(122), CKD(75), CKDM(9), CKL(75), CKLM(9),
1  CONDA(122), CONDD(10,9), CONDI(122), CONDJ(122), CP(75), CPMAT(9),
2  DELTIN,DTIM(75),DTIME,DTMIN,EL(75),FLUXT(15),HCONO,IO(20),
3  IFL(3),IFLAY(5,3),IEND,IMAT(75),TMELT(75),INDUT,IPLOT,ISEG(75),
4  TSTOP,JFL(3),KCH,KCONDI(122),KCONDJ(122),KCPT(122),KCPJ(122),
5  KMAT(3,5),KSJR,LAYN(75),MCON,NGCONO,NEL,NELL(3,5),NELS(3),
6  NFLUX,NLAY(3),NMAT,NOPT,NPRINT,NSEG,NTSPEC,NWLA,NWLI,PO,
7  PRINT,Q(75),QAR(3),QCONO(75),QCONV(3),QDMAX,QFLU,QR(50,3),
8  QRA7,QREF(2,3),QRR(14),SEGL,T(75),TAM,TCUT,TEF(3),TEMP(75)
COMMON TELUX(15),TES,TIME,TLAY(3,5),TP(75),TR,TSPEC(3),TSTOP,
1  VEL,WAVE(50,3),WAVEL(14),WL(10,10),WT(75),X(75),XCON(75),
2  XSEG(4),XTP,Y(75),YSEG(4)
COMMON /BLOCK/ ALFAT(10,14,9),CONDL(10,9),CPM(10,9),FMT(10,9),
1  HOF(9),KLD(9),NAT(9),NCPT(9),NCT(9),NMATT,NWL,PI,PHOM(9),
2  TALE(10,9),TCOND(10,9),TCP(10,9),TMELT(3),WAVEB(15)

C      UPDATES CRITICAL TEMPERATURE RATIO IF KC.NE.0
C      SET UP MATERIAL CONDUCTIVITIES AND SPECIFIC HEATS

IIE=NEL
DO 30 IE=1,IU
IF (TMELT( IE).EQ.1) GO TO 30
NM=IMAT( IE)
TEL=TEMP( IE)
IF (TEL.GE.TMELT(NM)) GO TO 27
7 CALL INT17 (7,TEL,NCPT(NM),TCOND(1,NM),CONDL(1,NM),CKL(IE))
IF (KLD(NM).EQ.2) GO TO 10
CKI(IE)=CKL(IE)
GO TO 20
10 CALL INT17 (8,TEL,NCPT(NM),TCOND(1,NM),CONDD(1,NM),CKD(IE))
20 IF (TEL.LT.TMELT(NM)-10.0) GO TO 25
CR(IE)=0.1*HOF(NM)
GO TO 30
25 CALL INT17 (11,TEL,NCPT(NM),TCP(1,NM),CPM(1,NM),CP(IE))
GO TO 30
27 I=NCPT(NM)
CKL(IE)=CONDL(I,NM)
IF (KLD(NM).EQ.2) GO TO 24
CKD(IE)=CKL(IE)
GO TO 20
24 CKI(IE)=CONDD(I,NM)
29 I=NOPT(NM)
GO TO 22
30 CONTINUE

C      CALCULATE CONDUCTIVE HEAT FLOW

ICCU=NGCONO
DO 50 ICC=1,ICCU
I=KCONDI(ICC)
J=KCONDJ(ICC)
IF (TMELT( I)+TMELT( J).GT.0) GO TO 40
IF (KCPJ( ICC).EQ.2) GO TO 40
CKI=CKL(I)
GO TO 50
40 CKI=CKD(I)
50 IF (KCPJ( ICC).EQ.2) GO TO 60
CKJ=CKL(J)
GO TO 70
60 CKJ=CKD(J)
70 CKIJ=CONDI( ICC)/CKI+CONDJ( ICC)/CKJ+CONDA( ICC)
CC=(TEMP( I)-TEMP( J))/CKIJ
QCONDI( I)=QCONDI( I)-CC
QCONDJ( J)=QCONDJ( J)+CC
Q( I)=Q( I)-CC
Q( J)=Q( J)+CC
80 CONTINUE

```

DTEMP


```

0      UPDATE TEMPERATURES AND MELT CONDITIONS
      IF (KC.EQ.0) GO TO 140
      DO 110 IF=1,IJ
      IF (TMELT(  IE).EQ.1) GO TO 110
      NM=IMAT(IE)
      TM=TMELT(NM)-13.0
      TEL=TEMP(IE)
      TEMP(  IE)=TEL+Q(  IE)*DELTIM/(WT(  IE)*CP(IE))
      IF (TEL.GT.TM) GO TO 104
      IF (TEMP(  IE).LE.TM) GO TO 105
      QQ=Q(  IE)-(TM-TEL)*WT(  IE)*CP(IE)/DELTIM
      TEMP(  IE)=TM+QQ*DELTIM/(WT(  IE)*0.1*HOF(NM))
104    IF (TEMP(  IE).LT.*MELT(NM)) GO TO 105
      IS=ISEG(  IE)
      IF (IE.NE.IEL(IS)) GO TO 105
      TMELT(  IS)=1
      IEL(IS) = IEL(IS) + 1
      NELS(IS) = NELS(IS) - 1
      IL = LAYN(IE)
      NELL(IS,IL) = NELL(IS,IL) - 1
      TEMP(  IS)=0.0
      Q(  IE)=0.0
      IF (NELS(  IS).GT.0) GO TO 110
      IEL(IS) = 0
      JEL(IS) = 0
105    CONTINUE
110    CONTINUE
0      140 RETURN
      END

```

DTEMP


```

0000 TIME .GT. 0.
1100 DO 1200 J=JL,NFLUX
      IF (TFLUX(J).GT.T) GO TO 1300
1200 CONTINUE
      JL = NFLUX
      Q00 = FLUXT(JL)
      GO TO 1400
1300 JL = J
      Q1 = FLUXT(J-1)
      T1 = TFLUX(J-1)
      Q00 = Q1 + (T-T1)*(FLUXT(J)-Q1)/(TFLUX(J)-T1)
1400 GET JPN
      END

```

FLUX

00000000

SUBROUTINE INT1Z (ICODE,X,NX,XT,YT,R)

SUBROUTINE INT1 IS A LINEAR INTERPOLATION ROUTINE.
 GIVEN A VALUE X, INT1 RETURNS THE CORRESPONDING VALUE Y.
 ICODE - CODE FROM CALLING PROGRAM FOR TRACEBACK IN CASE OF ERROR.
 NX = DIMENSION OF X-TABLE AND Y-TABLE IN CALLING PROGRAM.
 XT = TABLE OF X-VALUES IN CALLING PROGRAM.
 YT = TABLE OF Y-VALUES IN CALLING PROGRAM.
 R = RETURN VALUE Y.

```

    DIMENSION XT(1),YT(1)
    IF (X-XT(1))1,8,2
1  WRITE(6,11) X,(XT(I),I=1,NX)
    GO TO 13
2  IF (X-XT(NX)) 4,9,3
3  WRITE(6,11) X,(YT(I),I=1,NX)
    GO TO 10
4  DO 5 I=2,NX
    IF (Y-XT(I)) 7,6,5
5  CONTINUE
6  R=YT(I)
    RETURN
7  R=YT(I-1)+(X-XT(I-1))*(YT(I)-YT(I-1))/(XT(I)-XT(I-1))
    RETURN
8  R=YT(1)
    RETURN
9  R=YT(NX)
    RETURN
10 WRITE (6,12) (YT(I),I=1,NX)
    WRITE (6,14) ICODE
    STOP
  
```

0 FORMAT STATEMENTS -

```

11 FORMAT (9H X-VALUE F12.5,1X,14HOUTSIDE TABLE./
1  1X,9HX-TABLE =./6(2X,E12.5))
12 FORMAT (9H Y-TABLE = /(1X,6E14.5))
14 FORMAT (33H0TRACEBACK ERROR CODE IN INT1Z = I4)
    END
  
```

INT1Z


```

SUBROUTINE RPLOT (KP)
THIS ROUTINE SETS UP THE PLOT FILES FOR TEMP VS TIME.
COMMON A(75),ALF(10,10),B(122),CKD(75),CKDM(9),CKL(75),CKLM(9),
1  CONDA(122),CONDD(10,9),CONDI(122),CONDJ(122),CP(75),CPMAT(9),
2  DELTIM,DTIM(75),DTIME,DTMIN,EL(75),FLUXT(15),HCONO,IJ(20),
3  IEL(3),IELAY(5,3),IEND,TMAT(75),TMELT(75),INOJT,IPLT,ISEG(75),
4  TSTOP,JEL(3),KCH,KCONDI(122),KCONDJ(122),KCP(122),KCPJ(122),
5  KMAT(3,5),KSUP,LAYN(75),MCOB,NCONOC,NEL,NELL(3,5),NELS(3),
6  VFUX,NLAY(3),NMAT,NOPT,NPOINT,NSEG,NTSPEC,NALA,NWLT,PO,
7  PRINT,Q(75),QAB(3),QCONO(75),QCONV(3),QOMAX,QFLU,QP(50,3),
8  QRAD,QREF(2,3),QRP(14),SEGL,T(75),TAM,TCUT,TEF(3),TEMP(75),
COMMON TFLUX(15),TFS,TIME,TLAY(3,5),TP(75),TR,TSPEC(3),TSTOP,
1  VEL,WAVE(50,3),WAVEL(14),WL(10,10),WT(75),X(75),XCON(75),
2  YSEG(4),XTP,Y(75),YSEG(4)
DIMENSION TEM(5)
NPOINT = MAXIMUM NUMBER OF POINTS TO BE PLOTTED.
DATA NPOINT/200/

IF (KP-1) 100,400,800

100 NPLT = 0
N = (TSTOP + DELTIM)/DELTIM
NUPLT = 2*N/NPOINT
IF (NUPLT.EQ.0) NUPLT = 1
SET UP PLOT FOR EACH SURFACE AND LAYER BOUNDARY.
DO 300 IS=1,NSEG
ILU = NLAY(IS)
IELAY(1,IS) = IEL(IS)
DO 200 IL=1,ILU
200 IELAY(IL+1,IS) = IELAY(IL,IS) + NELL(IS,IL)
300 IELAY(ILU+1,IS) = IELAY(ILU+1,IS) - 1
IS = 1
IF (NSEG.GT.1) IS = 2
ILU = NLAY(IS)
ILU1 = ILU + 1
NSP IS THE NUMBER OF SEGMENTS PLOTTED.
NSP = 1
WRITE (4) (ID(I),I=1,20),QOMAX,TCUT,ILU1,NOPT,NSP
NPTS = 0
RETURN

400 IF (NPLT.GT.0) GO TO 700
NPLT = NUPLT
NPTS = NPTS + 1
OUTER SURFACE.
IE = IELAY(1,IS)
TE(1) = TEMP(IE)
IF (TEM(1).EQ.0.0) GO TO 440
IF (NELL(IS,1).LE.1) GO TO 440
TE(1) = TEMP(IE) + 0.5*(TEMP(IE) - TEMP(TE+1))
INNER SURFACE.
440 IE = IELAY(ILU1,IS)
TE(1) = TEMP(IE)
IF (TEM(1).EQ.0.0) GO TO 480
IF (NELL(IS,ILU1).LE.1) GO TO 480
TE(1) = TEMP(IE) + .5*(TEMP(IE) - TEMP(IE-1))
LAYER BOUNDARIES.
480 IF (ILU.EQ.1) GO TO 650
DO 500 IL=2,ILU
TE = IELAY(IL,IS)
TA = TEMP(IE-1)
IF (TA.EQ.0.0) GO TO 520
T1 = TA
IF (NELL(IS,IL-1).LE.1) GO TO 520
TA = TA + .5*(TA - TEMP(IE-2))
520 TR = TEMP(TE)
IF (TR.EQ.0.0) GO TO 540
T2 = TR
IF (NELL(IS,IL).LE.1) GO TO 560
TR = TR + .5*(TR - TEMP(IE+1))
560 IF (TA*TR.EQ.0.0) GO TO 580
TE = 0.5*(TA + TR)
IF (T2.GT.T1) GO TO 565

```

RPLOT

```

      TE = AMAX1(TE,T2)
      TE = AMIN1(TE,T1)
      GO TO 570
565  TE = AMAX1(TE,T1)
      TE = AMIN1(TE,T2)
570  TE4(IL) = TE
      GO TO 600
580  TE4(IL) = AMAX1(TA,TR)
600  CONTINUE
650  CONTINUE
      WRITE (8) TIME, (TEM(I),I=1,ILU1)
700  NPLOT = NPLOT + 1
      RETURN
C
800  ENDTITLE 8
      WRITE (6,1000)  ILU1,NPTS
      IFLOT = NPTS
1000  FORMAT (30HNUMBER OF CURVES GENERATED = I4/
1      30HNUMBER OF POINTS PER CURVE = I4)
      RETURN
END

```

RPL0T

```

      SUBROUTINE RPPRINT
      COMMON A(75),ALF(10,10),R(122),OKD(75),OKDM(9),OKL(75),OKLM(3),
1  CONDA(122),CONDD(10,9),CONDI(122),CONDJ(122),CP(75),CONMAT(2),
2  DFLTIM,DTIM(75),DTIME,DTMIN,EL(75),FLUXT(15),HCONO,IQ(20),
3  IEL(3),IFLAY(5,3),IEND,IMAT(75),IMELT(75),INJIT,IPLT,TSEG(75),
4  ISTOP,JEL(3),KCH,KCONDI(122),KCONDJ(122),KOPI(122),KOPJ(122),
5  KMAT(3,5),KSJR,LAYN(75),MCOO,NCONDO,NEL,NELL(3,5),NELS(3),
6  NFLUX,NLAY(3),NYAT,NOPT,NPPRINT,NSEF,NTSPEC,NWLA,NWLT,PO,
7  PRINT,Q(75),QAR(3),QCONO(75),QCONV(3),QOMAX,QFLU,QP(50,3),
8  QPAD,QPER(2,3),QRP(14),SEGL,T(75),TAM,TOUT,TEF(3),TEMP(75)
      COMMON TFLUX(15),TFS,TIME,TLAY(3,5),TP(75),TP,TSPEC(3),TSTOP,
1  WFL,WAVE(50,3),WAVEL(14),WL(10,10),WT(75),X(75),XCON(75),
2  XSEG(4),XTR,Y(75),YSEG(4)

```

```

      WRITE (6,1000) TIME,QPAD,QFLU
      WRITE (6,1100)
      DO 30 IE=1,NFL
      IS = ISEG(IE)
      IE1 = IEL(IS)
      IE2 = JEL(IS)
      IF (IE.EQ.IE1) GO TO 50
      IF (IE.EQ.IE2) GO TO 60
      WRITE (6,1200) IS,IE,TEMP(IE),QCONO(IE),Q(IE)
      GO TO 90
50 IF (IE.EQ.IE2) QPER(1,IS) = QPER(1,IS) + QPER(2,IS)
      WRITE (6,1300) IS,IE,TEMP(IE),QAR(IS),QPER(1,IS),
1  QCONV(IS),QCONO(IE),Q(IE)
      GO TO 90
60 WRITE (6,1400) IS,IE,TEMP(IE),QPER(2,IS),QCONO(IE),Q(IE)
      CONTINUE
      RETURN

```

```

1000 FORMAT (13H0TIME, SEC = E14.6/
1  34H INCIDENT FLUX, BTU/IN**2/SEC = E14.6/
2  34H INCIDENT FLUENCE, BTU/IN**2 = E14.6)
1100 FORMAT (31H0 SEGMENT ELEMENT TEMPERATURE,9X,
1  37H E A T F L O W , BTU / S E C /
2  23H NUMBER NUMBER DEGREES R,4X,
3  43H ABSORPTION RADIATION CONVECTION CONDUCTION,6X,3H"FT")
1200 FORMAT (1H ,I6,6X,I2,4X,E11.4,41X,E11.4,2X,E11.4)
1300 FORMAT (1H ,I6,6X,I2,4X,E11.4,5(2X,E11.4))
1400 FORMAT (1H ,I6,6X,I2,4X,E11.4,15X,E11.4,13X,2(2X,E11.4))
      END

```

2000 INT

```

SUBROUTINE SETUP
COMMON A(75),ALF(10,10),R(122),CKD(75),CKDM(9),CKL(75),CKLM(9),
1  CONDA(122),CONDD(10,9),CONDI(122),CONDJ(122),CP(75),CPMAT(9),
2  DELTIM,DTIM(75),DTIME,DTMIN,EL(75),FLUX(15),HCONO,IO(20),
3  IEL(3),IELAY(3,3),IEND,IMAT(75),IMELT(75),INOUT,IPLT,ISEG(75),
4  ISTOP,JFL(3),KCH,KCONDI(122),KCONDJ(122),KCP(122),KCPJ(122),
5  KMAT(3,5),KSUP,LAYN(75),MCOB,NCONDC,NEL,NELL(3,5),NELS(3),
6  NFLUX,NLAY(3),NMAT,NOPT,NPRINT,NSEG,NTSPEC,NWLA,NWLI,PO,
7  PRINT,Q(75),QAB(3),QCONO(75),QCONV(3),QDMAX,QFLU,QR(50,7),
8  QRAO,QRR(2,3),QRR(14),SEGL,T(75),TAM,TCUT,TEF(3),TEMP(75)
COMMON TELUX(15),TFS,TIME,TLAY(3,5),TP(75),TR,TSPEC(3),TSTOP,
1  VEL,WAVE(50,3),WAVEL(14),WL(10,10),WT(75),X(75),XCON(75),
2  XSEG(4),XTP,Y(75),YSEG(4)
COMMON /BLOCK/ ALFAT(10,14,9),CONDL(10,9),CPM(10,9),EMI(10,9),
1  JOF(9),KLD(9),MAT(9),NCF(9),NCT(9),NMATT,NWL,PI,PHCN(9),
2  TALE(10,9),TCO(10,3),TCP(10,9),TMELT(9),WAVEB(15)
DIMENSION PRCOR(8),PRKCH(22)
DATA PRCOR/10ALUMINUM,10HALUMINUM,10HMAGNESIUM,10HTITANIUM,
1  1042024-T3,10H7075-T6,10H A7318-H24,10HTI-8MN /
DATA PRKCH/10HABSORPTIVI,10HCONDUCTIVI,10HTHERMAL EX,10HSPECIFIC H
1  10HMELTING TE,10HHEAT OF FU,10HMODULUS OF,10HTENSILE YI,
2  10HTENSILE UL,10HULTIMATE S,10HDENSITY,2HTY,2HTY,7HPANSION,
3  3HEAT,10HTEMPERATURE,4HSION,10H ELASTICITY,10HFELD STRESS,
4  10HTIMATE STR,5HTRAIN,1H /
DATA IEN1/4HEND /, IEN2/4 /

```

```

1  FORMAT (6I12)
2  FORMAT (6F12.1)
3  FORMAT (20A4)
ISTOP=0
READ(5,3) (ID(I),I=1,20)
IF(ID(1).NE.IEN1.OR.ID(2).NE.IEN2) GO TO 4
ISTOP=1
GO TO 899
4  READ(5,1) INOUT,IPLT
IF (INOUT.GT.0) WRITE (6,951) (ID(I),I=1,20),IPLT
951  FORMAT (1H1,20A4//30H PLOT CODE (0 FOR NO PLOTS) = I2)
DO 5 I=1,NWL
5  WAVEB(I)=.5*(WAVEB(I)+WAVEB(I+1))
XMIN=10.0F10
XMAX=-10.0F10
IK=1
NEL=0
NMAT=NMATT
KSUP=1
NS=1
READ(5,1) NSEG
ISU=NS+1
DO 13 IS=1,ISU
READ(5,2) XSEG(IS),YSEG(IS)
XMIN=AMIN1(XMIN,XSEG(IS))
13  XMAX=AMAX1(XMAX,XSEG(IS))
ISJ=ISU-1
DO 26 IS=1,ISU
READ(5,1) NLAY(IS)
ILU=NLAY(IS)
DO 14 IL=1,ILU
READ(5,2) TLAY(IS,IL)
READ(5,1) KMAT(IS,IL),NELL(IS,IL)
NMAT=MAX0(NMAT,KMAT(IS,IL))
14  CONTINUE

```

2 CALCULATE SEGMENT LENGTHS

```

DX=XSEG(IS+1)-XSEG(IS)
DY=YSEG(IS+1)-YSEG(IS)
SEGL=SQRT(DX**2+DY**2)
DX=DX/SEGL
DY=DY/SEGL
DXL=DX
DYL=DY

```

2 SET UP ELEMENT NUMBERING SYSTEM,CALCULATE ELEMENT THICKNESS,

SETUP


```

C      AREA, WEIGHT, LENGTH, POSITION
      IFL(TS)=NEL+1
      DO 25 IL=1,ILU
      THICK=TLAY(IS,IL)/FLOAT(NELL(IS,IL))
      AREA=THICK*SEGL
      IF (IL.GT.1) GO TO 22
      IET=IFL(TS)
      Y(IET)=0.5*(XSEG(IS)+YSEG(IS+1))+THICK*DXL
      Y(IET)=0.5*(YSEG(IS)+YSEG(IS+1))-THICK*DXL
22     IFL=NEL+1
      IELJ=NEL+NELL(TS,IL)
      IF(IELU.LE.75) GO TO 23
      WRITE (6,941) NS
      STOP
23     DO 24 IE=IFLL,IFLU
      LAYN(IE) = IL
      KMAT(IE)=KMAT(IS,IL)
      XSEG(IE)=IS
      Y(IE)=THICK
      EL(IE)=SEGL
      A(IE)=AREA
      IF(IE.EQ.IET) GO TO 24
      X(IE)=X(IE-1)+0.5*(T(IE-1)+T(IE))*DXL
      Y(IE)=Y(IE-1)+0.5*(T(IE-1)+T(IE))*DXL
24     CONTINUE
25     NEL=IFLU
      NELS(TS)=IELU+1-IFL(TS)
      JEL(TS) = NEL
26     CONTINUE
      IF (INOUT.EQ.0) GO TO 29
      WRITE (6,910) NSEG
      ISU = NSEG + 1
      WRITE (6,904) (XSEG(TS),IS=1,ISU)
      WRITE (6,905) (YSEG(TS),IS=1,ISU)
      ISU = ISU - 1
      DO 312 IS=1,ISU
      ILJ=NLAY(TS)
      WRITE (6,911) TS,ILU
      WRITE (6,913) (IL,TLAY(IS,IL),KMAT(IS,IL),NELL(IS,IL),IL=1,ILU)
912    CONTINUE

C      CALCULATE HEAT CONDUCTION CONSTANTS FOR ELEMENTS WITHIN BRANCH
29     DO 30 IS=1,ISU
      ISC=1
      IELLI=IFL(TS)
      IELLJ=IELLI-1+NELS(TS)
      IF (TS.LT.ISU) GO TO 31
      IF (XSEG(1).NE.XSEG(IS+1).OR.YSEG(1).NE.YSEG(IS+1)) GO TO 33
      IELLJ=IFL(1)
      IELLJ=IELLJ-1+NELS(1)
      GO TO 32
31     IELLJ=IFL(TS+1)
      IELLJ=IELLJ-1+NELS(TS+1)
32     DBI=0.0
      DTJ=0.0
      DBJ=T(IELLJ)
      IEJ=IELLJ
      GO TO 34
33     ISC=0
34     DO 35 IEI=IELLI,IELU
      IF (ISC.EQ.0) GO TO 38
      DTI=0.0
      DBI=DTI+T(IEI)
35     RTJ=AMIN1(DBI,DBJ)-AMAX1(DTI,DTJ)
      KCOND1(IK)=IEI
      KCONDJ(IK)=IEJ
      R(IK)=RTJ
      COND1(IK)=EL(IEI)/(2.0*RTJ)

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SETUP

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CONVJ(IK)=EL(IEJ)/(2.0*RIJ)
CONJAG(IK)=0.0
KCPJ(IK)=1
KCPJ(IK)=1
IK=IK+1
IF (DTJ.GT.DBI) GO TO 38
IF (IFJ.NE.IELUJ) GO TO 36
ISJ=0
GO TO 38
36 IEJ=IEJ+1
DTJ=DTJ+T(IEJ)
IF (DTJ.LT.DBI) GO TO 35
C ELEMENTS OF SAME SEGMENT
35 IF (IEI.EQ.IELUI) GO TO 40
KCONVI(IK)=IEI
KCONVD(IK)=IEI+1
RIJ=EL(IEI)
3(IK)=BIJ
CONVI(IK)=T(IEI)/(2.0*RIJ)
CONVJ(IK)=T(IEI+1)/(2.0*BIJ)
CONJAG(IK)=0.0
KCPJ(IK)=2
KCPJ(IK)=2
IK=IK+1
39 CONTINUE
40 CONTINUE
KCONVD = IK-1
C SET UP MATERIAL PROPERTIES TABLES FOR ADDED MATERIALS
NML=NMATT+1
IF (NML.GT.NMAT) GO TO 260
DO 250 NM=NML,NMAT
READ (5,1) MCOB
IF (MCOB.EQ.0) GO TO 115
IF (INOUT.EQ.0) GO TO 113
IF (MCOB.GT.NMATT) GO TO 112
WRITE (6,935) NM,PRCOR(MCOB),PRCOR(MCOB+4)
GO TO 113
112 WRITE (6,935) NM,MCOB
113 READ (5,1) KCH
IF (KCH.EQ.1) GO TO 115
NAT(NM)=NAT(MCOB)
NTJ=NAT(NM)
DO 114 NT=1,NTJ
TAL=INT,NM)=TALF(NT,MCOB)
EMI(NT,NM)=EMI(NT,MCOB)
DO 114 NW=1,NWL
114 ALFAT(NT,NW,NM)=ALFAT(NT,NW,MCOB)
GO TO 120
115 READ (5,1) NAT(NM)
NTJ=NAT(NM)
IF (INOUT.EQ.1) WRITE (6,936) PRKCH(1),PRKCH(12)
IF (INOUT.EQ.1) WRITE (6,918) NAT(NM)
DO 116 NT=1,NTJ
READ (5,2) TALF(NT,NM),EMI(NT,NM)
READ (5,1) NWLA
DO 116 NW=1,NWLA
116 READ (5,2) WL(NM,NT),ALF(NW,NT)
DO 117 NW=1,NWLA
CALL INT17 (12,WAVEL(NW),NWLA,WL(1,NT),ALF(1,NT),ALFAT(NT,NW,NM))
117 CONTINUE
IF (INOUT.EQ.0) GO TO 118
WRITE (6,913) TALF(NT,NM),EMI(NT,NM),NWLA,(WL(NM,NT),ALF(NW,NT),NW
1=1,NWLA)
118 CONTINUE
120 IF (MCOB.EQ.0) GO TO 125
IF (KCH.LT.2) READ (5,1) KCH
IF (KCH.EQ.2) GO TO 125
NCT(NM)=NCT(MCOB)

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SETUP

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      KLD(NM)=KLD(MCOR)
      NTJ=NCT(NM)
      DO 122 NT=1,NTJ
      TCOND(NT,NM)=TCOND(NT,MCOR)
      CONDL(NT,NM)=CONDL(NT,MCOR)
      IF (KLD(NM).EQ.1) GO TO 122
      CONDO(NT,NM)=CONDO(NT,MCOR)
122  CONTINUE
      GO TO 130
125  READ (5,1) KLD(NM)
      READ (5,1) NCT(NM)
      NTJ=NCT(NM)
      DO 123 NT=1,NTJ
      READ (5,2) TCOND(NT,NM),CONDL(NT,NM),CONDO(NT,NM)
      IF (KLD(NM).EQ.1) CONDO(NT,NM) = CONDL(NT,NM)
123  CONTINUE
      IF (INOUT.EQ.0) GO TO 130
      WRITE (6,936) PRKCH(2),PRKCH(13)
      WRITE (6,920) KLD(NM),NCT(NM), (TCOND(NT,NM),CONDL(NT,NM),CONDO(NT,NM),NT=1,NTJ)
130  IF (MCOR.EQ.0) GO TO 145
      IF (KCH.EQ.0) GO TO 141
      IF (KCH.LT.4) READ (5,1) KCH
      IF (KCH.EQ.4) GO TO 145
141  NCPT(NM) = NCPT(MCOR)
      NTJ = NCPT(NM)
      DO 142 NT=1,NTJ
      TCD(NT,NM) = TCD(NT,MCOR)
      CPM(NT,NM) = CPM(NT,MCOR)
142  GO TO 150
145  READ (5,1) NCPT(NM)
      NTJ=NCPT(NM)
      DO 143 NT=1,NTJ
      READ (5,2) TCD(NT,NM),CPM(NT,NM)
      IF (INOUT.EQ.0) GO TO 150
      WRITE (6,936) PRKCH(4),PRKCH(15)
      WRITE (6,922) NCPT(NM), (TCD(NT,NM),CPM(NT,NM),NT=1,NTJ)
150  IF (MCOR.EQ.0) GO TO 155
      IF (KCH.EQ.0) GO TO 151
      IF (KCH.LT.5) READ (5,1) KCH
      IF (KCH.EQ.5) GO TO 155
151  TMELT(NM)=TMELT(MCOR)
      GO TO 160
155  READ (5,2) TMELT(NM)
      IF (INOUT.EQ.0) GO TO 160
      WRITE (6,936) PRKCH(5),PRKCH(16)
      WRITE (6,923) TMELT(NM)
160  IF (MCOR.EQ.0) GO TO 165
      IF (KCH.EQ.0) GO TO 161
      IF (KCH.LT.6) READ (5,1) KCH
      IF (KCH.EQ.6) GO TO 165
161  HOF(NM)=HOF(MCOR)
      GO TO 170
165  READ (5,2) HOF(NM)
      IF (INOUT.EQ.0) GO TO 170
      WRITE (6,936) PRKCH(6),PRKCH(17)
      WRITE (6,924) HOF(NM)
170  IF (MCOR.EQ.0) GO TO 215
      IF (KCH.EQ.0) GO TO 211
      IF (KCH.LT.11) READ (5,1) KCH
      IF (KCH.EQ.11) GO TO 215
211  PHOM(NM) = PHOM(MCOR)
      GO TO 220
215  READ (5,2) PHOM(NM)
      IF (INOUT.EQ.0) GO TO 217
      WRITE (6,936) PRKCH(11),PRKCH(22)
      WRITE (6,929) PHOM(NM)
217  READ (5,1) KCH
      IF (KCH.EQ.0) GO TO 220
      WRITE (6,213) N4
213  FOR IAT (R)HOMATERIAL PROPERTY CHANGES IMPROPERLY SPECIFIED FOR MAT
      IERIAL I2)

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SETUP

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C      DETERMINE EMISSIVITIES OF ADDED MATERIALS IF REQUIRED
220  IF (EMI(1,NM).GT.0.0) GO TO 250
      NTU=NAT(NM)
      DO 240 NT=1,NTU
      NP=1
      EMISS=0.0
      WL1=259000.0/TALF(NT,NM)
      DO 230 N=3,130
      FN=N
      ZETA=0.1*FN
      WL2=WL1/FN
      IF (WL2.LT.WL(1,NT).OR.WL2.GT.WL(NWLA,NT)) GO TO 228
      CALL INT1Z(4,WL2,NWLA,WL(1,NT),ALF(1,NT),AL)
      EMISS=EMISS+AL*ZETA**3/(EXP(7*ZETA)-1.0)
      GO TO 230
228  IF (NP.GT.0) GO TO 230
      NP=1
      WRITE (6,229) NM,NT
229  FORMAT (14HFOR MATERIAL ,I2,16H AT TEMPERATURE ,I2,
17H4. INTEGRATION FOR EMISSIVITY REQUIRES ABSORPTIVITIES OUTSIDE OF
2 RANGE PROVIDED,/44H ABSORPTIVITIES TAKEN AS 0 WHERE UNAVAILABLE)
230  CONTINUE
      EMT(NT,NM)=1.5*EMISS/PI**4
240  CONTINUE
250  CONTINUE
260  TEU=NEL
      DO 270 IE=1,IEU
      NM=TMAT(IE)
270  WT(IE)=A(IE)*RHOM(NM)
      DO 300 IF=1,IEU
300  IMELT(IF) = 0

C      SET UP DISTANCES FROM LEADING EDGE FOR CONVECTION
      TSJ=NSEG
      DO 480 IS=1,TSJ
      IE=IFL(TS)
      XCON(IE) = X(IE)
      IF1 = IE + 1
      IFJ = IF + NELS(IS) - 1
      IF (IEU.LT.IF1) GO TO 480
      DO 460 I=IE1,IEJ
460  XCON(I) = XCON(IE)
480  CONTINUE
      READ (5,2) VEL,PO,TAM,HCONO
      IF(INOUT.EQ.0) GO TO 500
      WRITE (6,952) VEL,PO,TAM
      IF(HCONO.GT.0.0) WRITE(6,953) HCONO
500  READ(5,1) NOPT
      IF(INOUT.GT.0) WRITE(6,954) NOPT
      READ(5,2) ODMAX
      IF(INOUT.GT.0) WRITE(6,955) ODMAX
      IF(NOPT.EQ.3) GO TO 550
      READ(5,2) TCUT
      IF(INOUT.GT.0) WRITE(6,956) TCUT
      GO TO 600
550  READ(5,1) NFLUX
      IF(INOUT.GT.0) WRITE(6,957)
      DO 580 I=1,NFLUX
      READ(5,2) TFLUX(I),FLUXT(I)
      IF(INOUT.GT.0) WRITE(6,958) I,TFLUX(I),FLUXT(I)
580  CONTINUE
      NTSPEC=1
600  READ(5,1) NWLI,NTSPEC
      IF(NWLI.EQ.0) GO TO 700
      IF(INOUT.GT.0) WRITE(6,959) NWLI,NTSPEC
      IF(NTSPEC.EQ.1) GO TO 650
      READ(5,2) (TSPEC(I),I=1,NTSPEC)
      IF(INOUT.GT.0) WRITE(6,960) (TSPEC(I),I=1,NTSPEC)
650  IF(INOUT.GT.0) WRITE(6,961)
      DO 580 I=1,NTSPEC
      DO 580 J=1,NWLI

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SETUP


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2543 (5,2) WA/E(J,I),OP(J,I)
TE(INGOUT,GT,0) WRITE(6,952) I,J,WAVE(J,I),OP(J,I)
643 CONTINUE
721 READ(5,2) DELTIM,ISTOP,PRINT
TE(INGOUT,GT,0) WRITE(6,949) DELTIM,TSTOP,PRINT
DTIME=0.0
WRITE(6,953)
TE=DEL
DO 444 IE=1,IEU
IS = TSEG(IS)
TE (TE,NE.TEL(IS)) GO TO 910
TL = 1
GO TO 420
900 IF (IMAT(IE).EQ.IMAT(IE-1).AND.T(IE).EQ.T(IE-1)) GO TO 920
TL = TL + 1
920 WRITE(6,902) IE,IS,IL,IMAT(IE),T(IE)
943 CONTINUE
903 RETURN

902 FORMAT (1X,I4, 11X,I4,11X,I4,12X,I4,11X,F7.4)
903 FORMAT (26H1 SUMMARY OF THERMAL MODEL/8H0ELEMENT,7X,7HSEGMENT,1X,
13Y,3H LAYER,10X,3H MATERIAL,7X,9H THICKNESS/
27H NUMBER, 9X,5H NUMBER,9X,6H NUMBER,11X,4H CODE,11X,
34H(TN)///)
904 FORMAT (16H X COORDINATE,IN/(1X,5E15.7))
905 FORMAT (16H Y COORDINATE,IN/(1X,5E15.7))
910 FORMAT (20H0NUMBER OF SEGMENTS ,I5)
911 FORMAT (15H0SEGMENT NUMBER,I3/28H NUMBER OF LAYERS IN SEGMENT/
1(20I5))
913 FORMAT (13H0 LAYER NUMBER ,5X,29H THICKNESS OF LAYER (TLAY),IN,5X,
1324 CODE DEFINING MATERIAL OF LAYER,5X,28H NUMBER OF ELEMENTS IN L
24Y3/I6X,I2,12X,E15.7,23X,I15,20X,I15))
913 FORMAT (44H NUMBER OF TEMPERATURES FOR ABSORPTIVITY,T5)
912 FORMAT (25H TEMPERATURE (TALF),DEG.R,E15.7,5X,16HEMISSIVITY (EMI),
1E15.7/23H NUMBER OF WAVE LENGTHS,I5/25H WAVE LENGTH (WL),MICRONS,
21X,1-4H ABSORPTIVITY (ALF)/(5X,E15.7,17X,E15.7))
920 FORMAT (5H CODE,T5,44H AND NUMBER OF TEMPERATURES FOR CONDUCTIVITY
1,T5/26H TEMPERATURE (TCOND),DEG.R,5X,36H LENGTHWISE (COND1),BTU/IN
2-SEC-DEG.R,5X,35H DEPTHWISE (COND0),BTU/IN-SEC-DEG.R,5X/
7(6X,E15.7,21X,E15.7,26X,E15.7))
922 FORMAT (14H NUMBER OF TEMPERATURES FOR SPECIFIC HEAT,I5/
1264 TEMPERATURE (TCR),DEG.R,10X,33H SPECIFIC HEAT (CPM),BTU/LB-DEG
2,R/(5X,E15.7,23X,E15.7))
923 FORMAT (34H MELTING TEMPERATURE (TMELT),DEG.P,2X,E15.7)
924 FORMAT (24H HEAT OF FUSION (HOF),BTU/LB,2X,E15.7)
929 FORMAT (25H DENSITY (RHOM),LBS-IN**3,2X,E15.7//)
935 FORMAT (23H0NEW MATERIAL NUMBER, I3/
1 50H MATERIAL TO WHICH NEW MATERIAL CORRESPONDS (MCOPI),
2 2A10)
9351 FORMAT (20H0NEW MATERIAL NUMBER, I3/
1 50H MATERIAL TO WHICH NEW MATERIAL CORRESPONDS (MCOPI),I3)
936 FORMAT (23H0MATERIAL PROPERTY TO BE CHANGED (KCI),2A10)
940 FORMAT (17H0TIME INFORMATION/
4 42H INTEGRATION TIME INTERVAL (DELTIM),SEC,2X,E15.7/
5264 STOP TIME (TSTOP),SEC,2X,E15.7/36H NUMBER OF TIME INTERV
1ALS (NINT),2X,E15.7)
941 FORMAT (34H1SECTION I2,36H CONTAINS MORE ELEMENTS THAN ALLOWED)
952 FORMAT (34H0VELOCITY OF AIR STREAM, FT/SEC = E15.6/
1 32H AMBIENT (2004) PRESSURE, PSI = E15.6/
2 34H AMBIENT (2004) TEMPERATURE, DEG. F = E15.6)
957 FORMAT (64H0CONVECTIVE HEAT TRANSFER COEFFICIENT (BTU/IN**2/SEC/0R
1I = E15.6)
954 FORMAT (21H0THERMAL FLUX OPTION I3)
955 FORMAT (25H0 MAXIMUM FLUX (BTU/IN**2/SEC) = E15.6)
956 FORMAT (23H0 FLUX CUTOFF AT T (SEC) = E15.6)
957 FORMAT (140,14X,4HTIME,13X,4HFLUX)
954 FORMAT (17,2E15.6)
953 FORMAT (44H0SPECTRAL DISTRIBUTION OF THERMAL SOURCE/
1224 NUMBER OF WAVE LENGTHS = I2/22H NUMBER OF TIMES = I2)
961 FORMAT (2040 TIMES (SEC) = 5E17.6)
961 FORMAT (140,5X,14T,3X,14I,2X,10H WAVELENGTH,8X,4HFLUX)
962 FORMAT (17,I4,4X,2E15.6)
END

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SETUP

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SUBROUTINE SPECT (IFIRST,NTSPEC,TSPEC,NWLI,WAVE,QR,NWL,
1 WAVEFL,WAVER,T,QRR)

DETERMINE SPECTRAL DISTRIBUTION OF SOURCE AS A FUNCTION OF
14 WAVELENGTHS BETWEEN 0.2 MICRONS AND 5.5 MICRONS.
IFIRST = 0, FIRST PASS.
1, ALL OTHER PASSES.
NTSPEC = NUMBER OF TIMES THE SPECTRAL DISTRIBUTION IS SPECIFIED.
TSPEC = TIMES AT WHICH THE SPATIAL DISTRIBUTION IS SPECIFIED.
NWLI = NUMBER OF INPUT TABLE ENTRIES DEFINING DISTRIBUTION. IF
0, USE BUILT-IN DISTRIBUTION FOR 6000 WATT TUNGSTEN LAMP.
WAVE = INPUT WAVELENGTHS.
QR = INPUT INTENSITIES.
NWL = NUMBER OF STANDARD WAVELENGTHS - PROBABLY 14.
WAVEFL = 14 STANDARD WAVELENGTHS.
WAVER = 15 WAVELENGTHS DEFINING BAND WIDTHS OF WAVEFL.
T = TIME.
QRR = RELATIVE INTENSITY FOR 14 WAVELENGTHS.

DIMENSION QR(50,3),WAVE(50,3),WAVEFL(15,3),QRR(15,3),
1 QPRX(3,14),QRR(14),TSPEC(3),C(5)
DIMENSION WAVEFL(14),WAVE(15)
DATA NWLX/15/, WAVEFL/0.2,0.375,0.625,0.875,1.125,1.375,1.625,
1 1.875,2.25,2.75,3.25,3.75,4.25,4.75,5.25, 30*0.0/
DATA QRRX/0.0,0.085,0.68,0.995,0.89,0.59,0.44,0.31,0.205,0.06,
1 0.055,0.037,0.02,0.015,0.0, 30*0.0/
DATA C/1.0,4.0,2.0,4.0,1.0/

IF (IFIRST.GT.0) GO TO 600
IF (NWLI.GT.0) GO TO 200
USE 6 KW TUNGSTEN LAMP.
NTSPEC = 1
NWLI = NWLX
DO 100 J=1,NTSPEC
DO 100 I=1,NWLI
WAVE(I,J) = WAVEFL(I,J)
100 QPR(I,J) = QRRX(I,J)
INTEGRATE, USING SIMPSON'S RULE, OVER EACH BANDWIDTH.
200 DO 300 J=1,NTSPEC
QTOT = 0.0
DO 400 I=1,NWL
X1 = WAVER(I)
X5 = WAVER(I+1)
DELX = (X5-X1)/4.
FF = 0.0
DO 300 K=1,5
CALL INT17(1,X1,NWLI,WAVE(1,J),QRR(1,J),F)
X1 = X1 + DELX
FF = FF + F*C(K)
300 QPR(I) = FF*(X5-X1)/12.
400 QTOT = QTOT + QPR(I)
DO 500 I=1,NWL
500 QPRX(J,I) = QPR(I)/QTOT
DO 550 I=1,NWL
550 QRR(I) = QPRX(1,I)
RETJPN

SPECTRAL DISTRIBUTION VARIES WITH TIME.
600 IF (NTSPEC.EQ.1) RETURN
IF (T.GT.TSPEC(NTSPEC)) GO TO 800
DO 700 I=1,NWL
CALL INT17(3,T,NTSPEC,TSPEC,QRRX(1,I),QRR(I))
700 CONTINUE
RETJPN

800 DO 900 I=1,NWL
900 QRR(I) = QPRX(NTSPEC,I)
RETJPN
END

```

SPECT

```

SUBROUTINE TIN
0000 READ IN INPUT DATA.
0010 DIMENSION AA(40)
0020 WRITE (6,1000)
0030 REWIND 5
0040 J = 1
0050 READ (1,1) (AA(I),I=1,40)
0060 IF (EOF(1)) 400,200
0070 WRITE (6,2000) J,(AA(I),I=1,40)
0080 WRITE (5,1) (AA(I),I=1,40)
0090 J = J + 1
0100 GO TO 100
0110
0120 400 WRITE (6,3000)
0130 REWIND 5
0140 RETJPN
0150
0160 1 FORMAT (40A2)
0170 1000 FORMAT (1H1,/,10X,15HT P A P - M L /
0180 1 15H0*** INPUT DATA /)
0190 2000 FORMAT (1X,I4,2X,40A2)
0200 3000 FORMAT (23H0*** END OF INPUT DATA)
0210 END

```

TTN


```

SUBROUTINE TSTEP
COMMON A(75), ALF(10,10), B(122), CKD(75), CKDM(9), CKL(75), CKLM(9),
1  CONDAG(122), CONDD(10,9), CONDI(122), CONDJ(122), CP(75), CPMAT(9),
2  DELTIM,DTIM(75),DTIME,DTMIN,EL(75),FLUXT(15),HCONO,IO(20),
3  IEL(3),IELAY(3,3),IFNO,IMAT(75),IMELT(75),INOJT,IPLDT,ISEG(75),
4  ISTOP,JEL(3),KCH,KCONDI(122),KCONDJ(122),KCPI(122),KCPJ(122),
5  KMAT(3,5),KSJ,LAYN(75),MCOR,NCONDC,NEL,NELL(3,5),NELS(3),
6  VFLUX,NLAY(3),NMAT,NOPT,NPRINT,NSEG,NTSPFC,NWL,NWLI,PO,
7  PRINT,Q(75),QAR(3),QCONO(75),QCONV(3),QOMAX,QFLU,QP(50,3),
8  QPAD,QREF(2,3),QRR(14),SEGL,T(75),TAM,TCUT,TEF(3),TEMP(75),
COMMON TFLUX(15),TFS,TIME,TLAY(3,5),TP(75),TR,TSPEC(3),TSTOP,
1  VEL,WAVE(50,3),WAVEL(14),WL(10,10),WT(75),X(75),XCON(75),
2  XSEGL(4),XTP,Y(75),YSEG(4)
COMMON /BLOCK/ ALFAT(10,14,9),CONDL(10,9),CPM(10,9),EMI(10,9),
1  HCF(9),KLD(9),NAT(9),NCPT(9),NCT(9),NMATT,NWL,PI,PHCM(9),
2  TALF(10,9),TCO(10,9),TCP(10,9),TMFLT(3),WAVEB(15)

T1=0.0
T2=0.0
IU=NEL
DO 5 IF=1,IU
NM=IMAT(IF)
T1=AMAX1(T1,TCONO(1,NM),TCP(1,NM))
5 T2=AMAX1(T2,TMELT(NM))
DO 10 IF=1,IU
10 OTIM( IF)=100.0
DO 20 NT=1,10
TEL=T1+(FLOAT(NT-1)/9.0)*(T2-T1)
DO 20 NM=1,NMAT
IF (TEL.LT.TCONO(1,NM).OR.TEL.LT.TCP(1,NM)) GO TO 20
IF (TEL.GT.TMELT(NM)) GO TO 15
CALL INT17 (13,TEL,NCT(NM),TCONO(1,NM),CONDL(1,NM),CKLM(NM))
IF (KLD(NM).EQ.2) GO TO 12
CKDM(NM)=CKLM(NM)
GO TO 13
12 CALL INT17 (5,TEL,NCT(NM),TCONO(1,NM),CONDD(1,NM),CKDM(NM))
13 CALL INT17 (6,TEL,NCPT(NM),TCP(1,NM),CPM(1,NM),CPMAT(NM))
GO TO 20
15 T=NCT(NM)
CKLM(NM)=CONDL(T,NM)
IF (KLD(NM).EQ.2) GO TO 17
CKDM(NM)=CKLM(NM)
GO TO 13
17 CKDM(NM)=CONDD(I,NM)
18 T=NOPT(NM)
CPMAT(NM)=CPM(I,NM)
20 CONTINUE
DO 25 IF=1,IU
25 Q( IF)=0.0
ICCI=NCONDC
DO 30 ICC=1,ICCI
I=CONDI(ICC)
J=CONDJ(ICC)
NM=IMAT(I)
IF (KCPI( ICC).EQ.2) GO TO 30
CKI=CKLM(NM)
GO TO 32
30 CKI=CKDM(NM)
32 NM=IMAT(J)
IF (KCPJ( ICC).EQ.2) GO TO 40
CKJ=CKLM(NM)
GO TO 42
40 CKJ=CKDM(NM)
42 OKTJ=CONDI( ICC)/CKI+CONDJ( ICC)/CKJ+CONDAG( ICC)
OC=1.0/OKTJ
Q( I)=Q( I)+OC
50 Q( J)=Q( J)+OC
DO 50 IF=1,IU
IF (Q( IF).EQ.0.0) GO TO 60
NM=IMAT( IF)
OTIM( IF)=AMIN1(OTIM( IF),WT( IF)*CPMAT(NM)*0.8/Q( IF))
60 CONTINUE
70 CONTINUE

```

TSTEP


```

      WRITE (6,110)
110  FORMAT (26H4TIME INTERVAL INFORMATION)
      IU=NF1
      WRITE (6,120) (DTIM(I), IE=1, IU)
120  FORMAT (1/ (14 10E13.5)) 44H4TIME INTERVAL REQUIRED BY EACH ELEMENT/
      DTMIN = DTIM(1)
      I=1
      IF (IU.EQ.1) GO TO 135
      DO 130 IE=2, IU
      IF (DTIM(IE).GE.DTMIN) GO TO 130
      DTMIN = DTIM(IE)
      I=IE
130  CONTINUE
135  WRITE (6,140) DTMIN,I
140  FORMAT (26H4MINIMUM TIME INTERVAL IS E13.5,
1  2TH SECONDS FOR ELEMENT I3)
150  CONTINUE
      DELTIM=DTMIN
      DTIME=DELTIM
      RETURN
      END

```

TSTEP

SUBROUTINE XHEAT (KR)

CALCULATES EXTERNAL RADIATION AND CONVECTION HEAT FLOWS
FOR EACH ELEMENT.

KR = 0, SETUP ONLY.

1, CONVECTION AND RERADIATION ARE CONSIDERED.

2, CONVECTION, RERADIATION, AND EXTERNAL RADIATION ARE USED.

```
COMMON A(75), ALF(10,10), R(122), CKD(75), CKDM(9), CKL(75), CKLM(9),
1  CONDDAG(122), CONDD(10,9), CONDI(122), CONDJ(122), CP(75), CPMAT(9),
2  DELTIM, DTIM(75), DTIME, DTMIN, EL(75), FLUXT(15), HCONO, ID(20),
3  IEL(3), IELAY(5,3), IEND, IMAT(75), IMELT(75), INOJT, IPLOT, ISEG(75),
4  ISTOP, JEL(3), KCH, KCONDI(122), KCONDJ(122), KCPT(122), KCPIJ(122),
5  KMAT(3,5), KSUR, LAYN(75), MCON, NCONDC, NEL, NELL(3,5), NELS(3),
6  NFLUX, NLAY(3), NMAT, NOPT, NPRINT, NSEG, NTSPEC, NHLA, NWLI, PO,
7  PRINT, Q(75), QAB(3), QCOND(75), QCONV(3), QDMAX, QFLU, QR(50,3),
8  QPAD, QSER(2,3), QRR(14), SEGL, T(75), TAM, TCUT, TFF(3), TEMP(75),
COMMON TFLUX(15), TFS, TIME, TLAY(3,5), TP(75), TR, TSPEC(3), TSTOP,
1  VEL, WAVE(50,3), WAVEL(14), WL(10,10), WT(75), X(75), XCON(75),
2  XSEG(4), XTR, Y(75), YSEG(4)
COMMON /BLOCK/ ALFAT(10,14,9), CONDL(10,9), CPM(11,9), EMI(10,9),
1  HOF(9), KLD(9), NAT(9), NCPT(9), NCT(9), NMATT, NWL, PI, RHOM(9),
2  TALF(10,9), TCOND(10,9), TCP(10,9), TMELT(9), WAVER(15)
```

IEU=NEL

DO 5 IE=1,IEU

QCOND(IE) = 0.

5 Q(IE) = 0.

DO 7 IS=1, NSEG

QCONV(IS) = 0.

QAB(TS) = 0.

QRR(1,IS) = 0.

7 QRR(2,IS) = 0.

IF (KR-1) 10, 30, 20

10 CALL SPECT (0, NTSPEC, TSPEC, NWLI, WAVE, QR, NWL, WAVEL, WAVES, TIME, QRR)

CALL FLUX (0, NOPT, NFLUX, TFLUX, FLUXT, QDMAX, TCUT, TIME, QRA)

CALL CONVEC (0, PO, TAM, VEL, TEMP, EL, XCON, HCONO, IE, TP, QCON)

RTURN

20 CALL SPECT (1, NTSPEC, TSPEC, NWLI, WAVE, QR, NWL, WAVEL, WAVES, TIME, QRR)

CALL FLUX (1, NOPT, NFLUX, TFLUX, FLUXT, QDMAX, TCUT, TIME, QRA)

30 ISJ = NSEG

DO 30 IS=1, ISJ

IF = JEL(IS)

IF (IE.EQ.0) GO TO 50

INNER SURFACE.

NM = IMAT(IE)

TEL = TEMP(IE)

IF (TEL.LT.TMELT(NM)) GO TO 42

I = NAT(NM)

EMISS = EMT(I, NM)

GO TO 43

42 CALL INT17 (14, TEL, NAT(NM), TALF(1, NM), EMI(1, NM), EMISS)

43 QRR(2, IS) = -(1.0E-14/3.0)*EL(IE)*EMISS*TEL**4

Q(IE) = Q(IE) + QRR(2, IS)

50 IE = IEL(IS)

IF (IE.EQ.0) GO TO 90

CALCULATE CONVECTIVE HEAT FLOW

CALL CONVEC (1, PO, TAM, VEL, TEMP, EL, XCON, HCONO, IE, TR, QCON)

QCONV(IS) = QCON

Q(IE) = Q(IE) + QCON

CALCULATE RADIATION FROM SURFACE ELEMENTS

NM = IMAT(IE)

TEL = TEMP(IE)

IF (TEL.LT.TMELT(NM)) GO TO 62

I = NAT(NM)

EMISS = EMT(I, NM)

GO TO 63

XHEAT

```

62 CALL INT17 (9,TFL,NAT(NM),TALF(1,NM),EMI(1,NM),EMISS)
63 QREP(1,IS) = -(1.0E-14/7.0)*EL(IF)*EMISS*TFL**4
Q(IF) = Q(IF) + QREP(1,IS)
IF (KPEQ.1) GO TO 90

```

C OPAQUE PANELS AND SECTIONS.

```

NM=IMAT(IF)
DO 85 IWL=1,NWL
IF (TFL.LT.TMFLT(NM)) GO TO 87
T=NAT(NM)
ALFA=ALFAT(I,IWL,NM)
GO TO 85
87 CALL INT1Z (10,TFL,NAT(NM),TALF(1,NM),ALFAT(1,IWL,NM),ALFA)
85 QAB(IS) = QAB(IS) + QRAD*QRR(IWL)*ALFA*EL(T)
Q(IF) = Q(IF) + QAB(IS)
90 CONTINUE
RETURN
END

```

XHEAT

APPENDIX B
PROGRAM LISTING OF
APLOT


```

0000 PROGRAM APLOT (OUTPUT,TAPE6=OUTPUT,TAPE8=514,TAPE40=514,PLOTS)
0001 THIS PROGRAM PLOTS THE DATA GENERATED BY TRAP-ML AND STOPED ON
0002 FILES TAPE8. THE OUTPUT PLOT TAPE IS TAPE39.
0003 PROGRAM APLOT WAS WRITTEN BY KAMAN AVIDYNE, SEPT., 1973.
0004 COMMON /STORE/ DAT(200,6), IDD(20), QD, TC, IL1, TIM(600)
0005 COMMON /KACOM/ NCPW, ICIN(2), ICOUT(1), INP(3), NGRAFS, NOGRAF, NCRT
0006 DIMENSION NPC(6), XMT(2), YMT(2), XT(3), YT(4), IT2(7), IT3(5), IPSS(6)
0007 DIMENSION IFMT(1)
0008 DATA NX/10/, NY/13/
0009 DATA XT/4HTIME,4H (SE,4HC) /
0010 DATA YT/4HTEMP,4H (D,4HEG R,4H) /
0011 DATA NT1/40/, NT2/25/, NT3/22/
0012 DATA IT2/4HPK,4H FLU,4HX, R,4HTU/I,4HN2/S,4HEC =,4H /
0013 DATA IT3/4HPULS,4H DU,4HRTI,4HON, ,4HSEC ,4H= /
0014 DATA IFMT/4HFS.3/, IC/4/, IR/5/
0015 DATA IPSS/1,3,5,6,4,2/
0016 DATA AL/5.0/
0017 DATA NDIMEN/200/
0018
0019 CALL PLOT (0.,0.,-3)
0020 CALL FIRST (100)
0021
0022 20 IPLT = 0
0023    READ (8) (IDD(I),I=1,20), QD,TC,IL1,NOP,NSP
0024    IF (EOF(8)) 900,40
0025 40 WRITE (6,7)
0026    IL = IL1 - 1
0027
0028 50 IPLT = IPLT + 1
0029    I = IPLT
0030    READ (8) TIM(I), (DAT(I,J),J=1,IL1)
0031    IF (EOF(8)) 120,70
0032 70 WRITE (6,8) TIM(I), (DAT(I,J),J=1,IL1)
0033    GO TO 50
0034
0035 120 NC = IL1
0036    IPLT = IPLT - 1
0037    ELIMINATE MELTED REGION.
0038    DO 200 J=1,NC
0039    DO 150 I=1,IPLT
0040      N = I
0041      IF (DAT(I,J).EQ.0.0) GO TO 180
0042 150 CONTINUE
0043      NPC(J) = IPLT
0044      GO TO 200
0045 180 NPC(J) = N-1
0046 200 CONTINUE
0047    SCALE THE TABLE.
0048
0049    N=IARS(NPC(1))
0050    CALL KALE (TIM,AL,N,1,XM,DX,DELX)
0051    CALL KALE (DAT,AL,N,1,YM,DY,DELY)
0052    IF (NC.EQ.1) GO TO 350
0053    XMIN = XM
0054    XMAX = XM+DX*AL
0055    YMIN = YM
0056    YMAX = YM+DY*AL
0057    NLOC = 1
0058    DO 300 TSC=2,NC
0059      N=IARS(NPC(TSC))
0060      NLOC = NLOC+NDIMEN
0061      CALL KALE (TIM,AL,N,1,XM,DX,DELX)
0062      CALL KALE (DAT(NLOC),AL,N,1,YM,DY,DELY)
0063      TEST = XM+DX*AL
0064      IF (YM.LT.XMIN) XMIN = XM
0065      IF (TEST.GT.XMAX) XMAX = TEST
0066      TEST = YM+DY*AL
0067      IF (YM.LT.YMIN) YMIN = YM
0068      IF (TEST.GT.YMAX) YMAX = TEST
0069 300 CONTINUE
0070    XMT(1) = XMIN
0071    XMT(2) = XMAX
0072    YMT(1) = YMIN

```

APLOT

```

      YMT(2) = YMAX
      CALL KALF(XMT,AL,2,1,XM,DX,DELX)
      CALL KALF(YMT,AL,2,1,YM,DY,DELY)

      DRAW THE AXES.

350  XAX = 0.0
      IF (YM.LT.0.0.AND.-YM/DY.LT.AL) XAX=-YM/DY
      CALL KAXIS (0.0,XAX,XT,-NX,AL,0.0,XM,DX,DELX,0)
      YAX=0.0
      IF (XM.LT.0.0.AND.-XM/DX.LT.AL) YAX=-XM/DX
      CALL KAXIS (YAX,0.0,YT,NY,AL,90.0,YM,DY,DELY,-1)

      DRAW TITLES.
      H = 0.1
      CALL KASYM (0.3,5.5,H,IDD,0.0,NT1,3)
      CALL KASYM (0.3,5.3,H,IT2,0.0,NT2,3)
      CALL KANUM (999.0,5.3,H,DD,0.0,IFMT,IC,IP)
      IF (NOP.EQ.3) GO TO 400
      CALL KASYM (0.3,5.1,H,IT3,0.0,NT3,3)
      CALL KANUM (999.0,5.1,H,TC,0.0,IFMT,IC,IQ)

      PLOT EACH CURVE.

410  NFR = IPLOT/5
      DO 500 I=1,NC
      K = NDIMEN*(I-1) + 1
      NP = NPC(I)
      IPS = IPSS(I)
      CALL KALINE (TIM(1),DAT(K),NP,XM,DX,YM,DY,NFR,IPS)
500  WRITE (6,2) I

      INDICATE MELT CONDITION.
      H = 0.1
      DO 600 J=1,NC
      IF (NPC(J).EQ.IPLOT) GO TO 600
      I = NPC(J)
      K = NDIMEN*(J-1) + 1
      X = (TIM(I)-XM)/DX
      Y = (DAT(K)-YM)/DY - .5*H
      CALL KASYM (X,Y,H,IDDUM,0.0,-7,3)
600  CONTINUE

      CHECK FOR LAST PLOT.
      IF (NOGRAF.EQ.1) GO TO 700
      WRITE (6,4) NOGRAF
      GO TO 800
700  WRITE (6,5)
800  NOGRAF=NOGRAF+1

      ADVANCE TO NEXT FRAME.
      CALL KAVANS (AL+3.5,0.0)
      GO TO 20

      CLOSE THE PLOTTING SYSTEM.

900  NOGRAF = NOGRAF - 1
      WRITE (6,6) NOGRAF
      CALL PLOTE (NMH)
      STOP

      FORMAT STATEMENTS.
2  FORMAT (6H CURVE I3,1X,7H PLOTTED/)
4  FORMAT (6X,7H*****I3,1X,22H GRAPHS PLOTTED.***** )
5  FORMAT (5X,34H*****FIRST GRAPH PLOTTED.***** )
6  FORMAT (5X,7H*****I3,1X,37H GRAPHS PLOTTED, SYSTEM CLOSED.***** )
7  FORMAT (13H1DATA PLOTTED/)
8  FORMAT (1H,5I5.6,5Y,6F15.6)

```

APLOT

3

END

APLOT

SUBROUTINE FIRST(M)

OPEN PLOT FILES FOR OFF-LINE PLOTS.

M = NUMBER OF GRAPHS TO BE PLOTTED.

COMMON/KACOM/NCPW,ICIN(2),ICOUT(1),JNP(3),NGRAFS,NOGRAF,NCRT
DIMENSION I(2),K(3)
DIMENSION PLOTID(3)

DATA I/4H(,444A1)/,J/4H(A4)/,K/4H(,4H18A4,4H) /,L/4/
DATA PLOTID/10H7299 BILL ,10HLEE PHONE ,10H 2721990
N=IAPS(M)
XMAX=10*N
YMAX=11.4
CALL PLTID3(PLOTID,XMAX,YMAX,1.0)

NCRT = FLAG FOR TYPE OF PLOTTING.
REQUIRED AT SOME INSTALLATIONS FOR PROPER FILM ADVANCE
AND ORIGIN RESET.

0 = ORIGIN CONTROLLED BY USER.
1 = ORIGIN MOVED UP AND OVER 2.0 INCHES FOR USE WITH
QUICK PLOT ROUTINES.

NCRT=0
IF (M.GT.0) NCRT=1

IF (M.GT.0) CALL KAVANS (3.0,1.0)

IBM 360 - AVCO.
CALCOMP MODEL 890 CATHODE RAY TUBE PLOTTER.
MAXIMUM PLOT AREA WITHOUT DISTORTION = 7.5X7.5 INCHES.

CALL IDERM2('AVIDYNE ', 'A5', 0, T, 39)

N=IAPS(M)

IF M IS LESS THAN ZERO, USER IS EXPECTED TO ADVANCE
FILM BEYOND AVCO ID FRAME WITH

CALL CALCMP(0.0,0.0,0.2)

IF USER CALLS KAVANS ORIGIN WILL BE MOVED UP 1.05 INCHES,
AND RIGHT 1.25 INCHES - INTENSITY WILL BE CHANGED BY -5.

CHECK FOR QUICK PLOT OR USER PLOTTING.

MOVE PEN TO RIGHT OF I.O. FRAME AND SET INTENSITY
TO A LOWER LEVEL FOR FASTER BEAM MOVEMENT.

IF (M.GT.0) CALL KAVANS(1.05,1.25)

SET COUNTERS.
NGRAFS = TOTAL NUMBER OF GRAPHS TO BE GENERATED.
NOGRAF = NUMBER OF NEXT GRAPH TO BE PLOTTED.

NGRAFS=N
NOGRAF=1
NCPW=L
ICIN(1)=I(1)
ICIN(2)=I(2)
ICOUT(1)=J
JNP(1)=K(1)
JNP(2)=K(2)
JNP(3)=K(3)

WRITE (5,1)
1 FORMAT (31H0PLOT ROUTINE APL0T INITIALIZED)

STOP

2

RETURN
END

FIRST

SUBROUTINE KALINE(XV,YV,N,XM,DX,YM,DY,K,KODES)

CALLING ARGUMENTS -

XV = X COORDINATES OF DATA POINTS TO BE PLOTTED. (REAL)
 YV = Y COORDINATES OF DATA POINTS TO BE PLOTTED. (REAL)
 N = NUMBER OF DATA POINTS TO BE PLOTTED. (INTEGER)
 A NEGATIVE N INDICATES A POINT BY POINT PLOT WITH
 NO CONNECTING LINES.
 XM = MINIMUM VALUE OF VARIABLE ON ARSCISSA. (REAL)
 DX = SCALE FACTOR (UNITS/INCH) FOR ARSCISSA. (REAL)
 YM = MINIMUM VALUE OF VARIABLE TO BE PLOTTED ON ORDINATE. (REAL)
 DY = SCALE FACTOR (UNITS/INCH) OF VARIABLE TO BE PLOTTED ON
 ORDINATE. (REAL)
 K = POSITIVE OR NEGATIVE CODE. (INTEGER)
 IF POSITIVE, A SPECIAL CHARACTER FROM THE SYMBOL TABLE IS
 TO BE PLOTTED. THE NUMBER OF THE CHARACTER IS GIVEN IN
 KODES(1).
 IF NEGATIVE, THE USER SUPPLIES THE PLOT CODE FOR HIS
 OWN SPECIAL CHARACTER IN KODES.
 KODES(1) = NUMBER OF WORDS DESCRIBING SYMBOL.
 KODES(2) = START OF SYMBOL.
 IF K = 0, (1) NO SPECIAL SYMBOL IS TO BE USED, AND
 (2) EVERY POINT IN THE TABLE IS TO BE PLOTTED.
 MAGNITUDE OF K INDICATES POINTS IN TABLE TO BE PLOTTED.
 K = 1, EVERY POINT.
 K = 2, EVERY SECOND POINT.
 K = 3, EVERY THIRD POINT.
 ETC.

DIMENSION XV(1),YV(1),KODES(1)

IF (N.EQ.0) RETURN

CHECK FOR CONTINUOUS LINE.

IF(N.LT.0) GO TO 20

DRAW CONTINUOUS CURVE.

X=(XV(1)-XM)/DX
 Y=(YV(1)-YM)/DY
 CALL PLOT(X,Y,3)
 DO 10 I=1,N
 X=(XV(I)-XM)/DX
 Y=(YV(I)-YM)/DY
 10 CALL PLOT(X,Y,2)

CHECK FOR SPECIAL SYMBOLS.

IF (K.EQ.0) GO TO 50

MOVE PEN TO FIRST POINT.

20 Y=(YV(1)-YM)/DY
 Y=(YV(1)-YM)/DY
 CALL PLOT(X,Y,3)
 M=IABS(N)

CHECK FOR USER SUPPLIED CODE.

IF (K.LT.0) GO TO 60

L=K
 IK=1
 KODE=-KODES(1)

30 IL=L
 DO 40 I=IL,M,L
 X=(XV(I)-XM)/DX
 Y=(YV(I)-YM)/DY

40 CALL KASYM(X,Y,0.1,KODES(IK),0.0,KODE,3)

LIFT PEN

KALINE

2
0000

```
50 CALL FLOT(X,Y,3)
   RETJPN
   USER SUPPLIED CODE.
50 KODE=-20-KODES(1)
   IK=2
   L=-K
   GO TO 30
END
```

KALINE

SUBROUTINE KALE(D,AL,NP,J,AMIN,DA,DEL)

SUBROUTINE KALE PROVIDES SCALING INFORMATION FOR
THE AXIS ROUTINE KAXIS.

CALLING ARGUMENTS -

INPUT -

D = ARRAY OF DATA POINTS TO BE EXAMINED. (REAL)
AL = AXIS LENGTH, INCHES. (REAL)
NP = NUMBER OF POINTS TO BE SCANNED IN ARRAY. (INTEGER)
J = JUMP CODE WHOSE MAGNITUDE DETERMINES
THE DATA TO BE SCALED.
J = 1, USE EVERY POINT.
J = 2, USE EVERY SECOND POINT.
J = 3, USE EVERY THIRD POINT.
ETC.
(INTEGER)

OUTPUT -

AMIN = STARTING VALUE FOR VARIABLE ON AXIS. (REAL)
DA = SCALE FACTOR, UNITS PER INCH, TO BE USED ON AXIS. (REAL)
DEL = UNITS PER TIC MARK. (REAL)
ALLOWABLE DIVISIONS - 1.0, 2.0, 5.0 * POWER OF TEN.

CAUTION - WHEN PLOTTING ON GRAPH PAPER, PLEASE NOTE THAT
THE NUMBER OF MAJOR TIC MARKS IS DETERMINED BY THE DATA.
UNLIKE OTHER PLOT PACKAGES (SUCH AS CALCOMP, ETC.)
THIS ROUTINE WILL NOT ALWAYS PRODUCE ONE MAJOR
TIC MARK PER INCH.

DIMENSION D(1)

OBTAIN TOTAL NUMBER OF POINTS IN ARRAY, N.

N=NP*J

AMIN = MINIMUM VALUE OF DATA IN ARRAY.
AMAX = MAXIMUM VALUE OF DATA IN ARRAY.

AMIN=D(1)

AMAX=AMIN

DO 20 I=1,N,J

T = TEST VALUE.

T=D(I)

IF (AMIN.LT.T) GO TO 10

AMIN=T

GO TO 20

10 IF (AMAX.GT.T) GO TO 20

AMAX=T

20 CONTINUE

IF (AMAX.EQ.AMIN) AMAX=1.001*AMIN

IF (AMAX.LT.AMIN) AMAX=0.99*AMIN

IF (AMAX.NE.AMIN) GO TO 25

WRITE (6,1) AL,NP

1 FORMAT (64H1***POSSIBLE ERROR DETECTED BY KALE. DATA TO BE SCALED
1 ALL ZERO. /15H AXIS LENGTH = E13.5,1H,14.8H POINTS.)

SPECIAL DEBUG OUTPUT FOR CDC 6600.

I=LOC(D)

WRITE (6,2) I

2 FORMAT (28H STARTING ADDRESS OF DATA = 08/I)

AMAX=1.0

MINIMUM AND MAXIMUM SHOULD FALL ON TIC MARKS.

ALLOWABLE TIC MARKS = 1, 2, OR 5.

R = RANGE.

KALE


```

25 R=AMAX-AMIN
   POL=P/AL
   DUM=ALOG10(POL)
   I=INT(DUM)
   POL=POL/(10.0**I)
   IF (POL.GT.1.0) GO TO 30
   DEL=10.0**I
   GO TO 50
30 IF (POL.GT.2.0) GO TO 40
   DEL=2.0*10.0**I
   GO TO 50
40 IF (POL.GT.5.0) GO TO 45
   DEL=5.0*10.0**I
   GO TO 50
45 DEL=10.0**(I+1)
50 DUM=AMIN/DEL
   I=INT(DUM)
   IF (FLOAT(I).GT.DUM) I=I-1
   AMIN=FLOAT(I)*DEL
   R=AMAX-AMIN
   DUM = 1.0001*DEL*AL + AMIN
   IF (AMAX.LT.DUM) AMAX = DUM

NT = NUMBER OF TIC MARKS BEYOND STARTING VALUE.

NT=INT(R/DEL-0.0001)+1
IF (NT.GT.INT(AL)) GO TO 25
QA=DEL*FLOAT(NT)/AL
PRINT
END

```

KALE

SUBROUTINE KANUM(X,Y,H,F,ANGLE,IFMT,NC,NP)

SUBROUTINE KANUM PLOTS NUMBERS ACCORDING TO FORMAT IFMT.

CALLING ARGUMENTS -

X,Y = COORDINATES OF LOWER LEFT CORNER OF NUMBER
TO BE PLOTTED, IN INCHES, FROM THE ORIGIN. (REAL)
H = HEIGHT, IN INCHES, OF PLOTTED NUMBER.
F = NUMBER TO BE PLOTTED.
CAN BE EITHER INTEGER OR DECIMAL.
ANGLE = ANGULAR ORIENTATION, IN DEGREES, OF PLOTTED NUMBER
MEASURED COUNTERCLOCKWISE FROM HORIZONTAL AXIS. (REAL)
IFMT = FORMAT UNDER WHICH NUMBER, F, IS TO BE WRITTEN.
ANY ALLOWABLE FORTRAN FORMAT THAT CORRESPONDS
IN TYPE TO F. (HOLLERITH)
NC = NUMBER OF CHARACTERS IN FORMAT, IFMT. (INTEGER)
NP = NUMBER OF CHARACTERS RESULTING FROM FORMAT. (E.G. E16.4
RESULTS IN 16 CHARACTERS.) (INTEGER)

COMMON/KACOM/NCPW,ICIN(2),ICOUT(1),INP(3),NGPAFS,NOGRAF,NCRT
DIMENSION NFMT(18),IFMT(1),NOUT(18)
DATA NFMT(1)/4H(/,NFMT(18)/4H) /

IF (NC.EQ.0) GO TO 20

N = NUMBER OF ALPHAMERIC WORDS IN IFMT.

N=NC/NCPW
IF (N*NCPW.LT.NC) N=N+1
J=2

FILL FORMAT ARRAY.

DO 10 I=1,N
NFMT(J)=IFMT(I)
10 J=J+1
IF (J.LT.18) NFMT(J)=NFMT(18)

CONVERT TO ALPHAMERIC.

WRITE (40,NFMT) F
REWRITE 40
N=NR/NCPW
IF (N*NCPW.LT.NP) N=N+1
READ (40,TNP) (NOUT(I),I=1,N)
REWRITE 40

PLT CODE.

CALL KASYM(X,Y,H,NOUT,ANGLE,NP,3)
20 RETURN
END

KANUM

SUBROUTINE KASYM(X3,Y3,HEIGHT,IBCD,ANGLE,NCHAR,IPENC)

SUBROUTINE KASYM DRAWS ALPHANUMERIC INFORMATION
AND SPECIAL SYMBOLS AT ANY ANGLE AND ANY SIZE
ON THE PLOTTING AREA.

IN ITS STANDARD FORM SYMBOL IS USED TO PRINT TEXT
MATERIAL ALONG AXES, AND TITLE GRAPHS. THE LETTERS
A THROUGH Z, DIGITS 0 THROUGH 9 AND ALL OTHER STANDARD
FORTRAN CHARACTERS ARE AVAILABLE.

IN AN ALTERNATE FORM, IT WILL PLOT A CENTERED SYMBOL
FROM THE SPECIAL SYMBOL TABLE, OR A USER SUPPLIED SYMBOL.

THE ARGUMENT NCHAR CONTROLS THE FORM TO BE USED.

ARGUMENT LIST -

X3, Y3 = COORDINATES, IN INCHES, OF EITHER
(1) THE LOWER LEFT CORNER OF THE
FIRST CHARACTER OF STANDARD TEXT TO
BE PRINTED - OR -
(2) THE CENTER OF THE SPECIAL SYMBOL
TO BE PLOTTED.

HEIGHT = HEIGHT, IN EITHER INCHES OR CENTIMETERS,
OF THE TEXT CHARACTERS, OR SPECIAL SYMBOLS
TO BE PLOTTED.
IF HEIGHT IS GREATER THAN 0.0, THE UNITS
ARE ASSUMED TO BE IN INCHES.
IF HEIGHT IS LESS THAN 0.0, THE UNITS
ARE ASSUMED TO BE IN CENTIMETERS.

IBCD = TEXT MATERIAL TO BE PRINTED, OR
SPECIAL SYMBOL CODE, DEPENDING UPON
THE VALUE OF NCHAR.
STANDARD ALPHANUMERIC TEXT (BCD OR
A-TYPE FORMAT) SHOULD BE LEFT-JUSTIFIED
AND CONTIGUOUS IN
(1) A SINGLE VARIABLE,
(2) AN ARRAY, OR
(3) A FOLLERITH LITERAL (IF THE COMPILER PERMITS).

SPECIAL SYMBOL CODES SHOULD BE INTEGER FORMAT.

ANGLE = ANGULAR ORIENTATION, IN DEGREES, AT WHICH CHARACTER
LINE IS TO BE PRINTED. POSITIVE VALUES ARE MEASURED
COUNTERCLOCKWISE FROM THE HORIZONTAL X-AXIS, AND
NEGATIVE VALUES ARE MEASURED CLOCKWISE.

NCHAR = CONTROL CONSTANT FOR TYPE OF MATERIAL TO BE
PRINTED ON PLOT.

A POSITIVE NCHAR INDICATES STANDARD
ALPHANUMERIC TEXT IS TO BE PRINTED. NCHAR
IS THE NUMBER OF CHARACTERS TO BE PRINTED.

A NEGATIVE NCHAR INDICATES A SPECIAL
CENTERED SYMBOL IS TO BE PLOTTED. IF THE
ABSOLUTE VALUE OF NCHAR IS BETWEEN 1 AND 14,
A SPECIAL SYMBOL FROM THE TABLE IS PRINTED.
IF THE ABSOLUTE VALUE OF NCHAR IS GREATER
THAN 14, THE USER SUPPLIES HIS OWN
SPECIAL SYMBOL IN IBCD.
IF MORE THAN TWO WORDS ARE REQUIRED FOR THE
DEFINITION OF THE NEW SYMBOL, EXPAND
THE DIMENSION OF NCHAR.
THE DIMENSION OF THE NEW SYMBOL IN THE
CALLING PROGRAM IS NOW ASSUMED TO BE TWO.

IPENC = PEN CODE.
= 2, PEN DOWN DURING MOVE TO (X3,Y3).
= 3, PEN UP DURING MOVE TO (X3,Y3).

KASYM


```

H=HEIGHT
CHANGE CENTIMETERS TO INCHES IF NECESSARY.
IF (H.LT.0.0) H=-H*2.54
DEL = DISTANCE IN INCHES BETWEEN NODES.
DEL=H/7.0
DELC=DEL*COST
DELS=DEL*SINT
ISTART=1

DETERMINE TYPE OF CHARACTERS TO BE PLOTTED.
IF (NCHAR.GT.0) GO TO 70
SPECIAL SYMBOL.
N=-NCHAR
X2=X1+H*COST
Y2=Y1+H*SINT
IF (N.GT.NSYM) GO TO 60
SYMBOL IS CONTAINED IN TABLE.
10 INDEX=IN(N)
ICHAO=NODES(INDEX)
IPART=1
PLOT NODES.
20 NODE=ICHAO/100
IF (IPART.EQ.2) NODE=ICHAO-100*NODE
IF NODE = 40, LEFT PEN AND GO BACK FOR ANOTHER POINT.
IF (NODE.EQ.LEFT) GO TO 50
IF NODE = 44, END OF THIS SYMBOL.
IF (NODE.EQ.NEXT) GO TO 170
SEPARATE NODE INTO (X,Y) COORDINATES.
IX=NODE/10
IY=NODE-10*IX
IF (IX.GT.4) IX=4-IX
IF (IY.GT.4) IY=4-IY
OBTAIN COORDINATES OF NODE.
X=X1+DELC*FLOAT(IX)-DELS*FLOAT(IY)
Y=Y1+DELC*FLOAT(IY)+DELS*FLOAT(IX)
PLOT POINT.
IF (ISTART.EQ.1) CALL PLOT(X,Y,3)
ISTART=ISTART+1
IPEN=2
30 CALL PLOT (X,Y,IPEN)
OBTAIN NEXT POINT.
IPART=IPART+1
IF (IPART.LE.2) GO TO 20
INDEX=INDEX+1
GO TO 10
50 IPEN=3
ISTART=1
X=X1
Y=Y1
GO TO 30

```

KASYM

```

SPECIAL SYMBOL - USER SUPPLIED.
60 INDEX=NNODES+1
   J=N-20
   DO 65 I=1,J
   K=NNODES+I
65 NODES(K)=IRCD(I)

   GO TO 10

STANDARD FORTRAN CODE.
70 NW=1
   NC=1

   SHIFT TO CENTER OF GRID.
   X2=X1+3.0*(DELC-DELS)
   Y2=Y1+3.0*(DELC+DELS)
   DC7=7.0*DELC
   DS7=7.0*DELS
80 LETTEC=1
   IPART = 1

OBTAIN CODE OF CHARACTER.
WRITE (40,ICOUT)IRCD(NW)
READ 40
READ (40,ICIN) (ICHARS(I),I=1,NCPW)
READ 40
90 NODE = ICHARS(LETTER)
   DO 95 ITEM = 1,NMORN
   IF (NODE.EQ.NORM(ITEM)) GO TO 96
95 CONTINUE
WRITE (6,1) NODE,IRCD(NW)
1 FORMAT (66H0***WARNING***. NON-STANDARD FORTRAN SYMBOL ENCOUNTERED
1 BY KASYM. A1,2X,A10/)
RETURN

OBTAIN STARTING POINT IN TABLE.
96 INDEX = INORM(ITEM)

OBTAIN CHARACTER CODE FROM TABLE.
100 ICHAR=NORMAL(INDEX)
   IPART = 1

PLOT NODE.
110 NODE = ICHAR/100
   IF (IPART.EQ.2) NODE=ICHP-100*NODE

   IF NODE = 40, LIFT PEN AND GO BACK FOR ANOTHER POINT.
   IF (NODE.EQ.LIFT) GO TO 140
   IF NODE = 44, END OF THIS CHARACTER.
   IF (NODE.EQ.NEXT) GO TO 150

SEPARATE NODE INTO (X,Y) COORDINATES.
IX = NODE/10
IY = NODE-10*IX
IF (IY.GT.4) IY = 4-IY
IF (IY.GT.4) IY = 4-IY

PLOT NODE.
X=X2+DELC*FLOAT(IX)-DELS*FLOAT(IY)
Y=Y2+DELC*FLOAT(IY)+DELS*FLOAT(IX)

```

KASYM

```

      IF (ISTART.EQ.1) CALL PLOT(X,Y,3)
      ISTART=ISTART+1
      IPEN=2
120  CALL PLOT (X,Y,IPEN)
      OBTAIN NEXT POINT.
      IPART = IPART+1
      IF (IPART.LE.2) GO TO 110
      INDX=X=INDEX+1
      GO TO 100
140  IPEN=3
      X=X2
      Y=Y2
      ISTART=1
      GO TO 120
      NEXT CHARACTER.
150  LETTER=LETTER+1
      CALL PLOT(X2,Y2,3)
      NC=NC+1
      ISTART=1
      X2=X2+DC7
      Y2=Y2+DS7
      IF (NC.GT.NCHAR) GO TO 160
      IF (LETTER.LE.NCPW) GO TO 90
      INCREASE WORD COUNT.
      NW=NW+1
      GO TO 80
      RESET PEN AT NEW (X2,Y2).
160  XSAVE=X2 -3.0*(DELC -DELS)
      YSAVE=Y2 -3.0*(DELC +DELS)
165  CALL PLOT(X1,Y1,3)
      RETJPN
170  XSAVE=X2
      YSAVE=Y2
      GO TO 165
      END

```

KASYM

SUBROUTINE KAVANS(X,Y)

SUBROUTINE KAVANS ADVANCES THE FILM TO THE NEXT FRAME, AND
RESETS THE LOGICAL ORIGIN PREVIOUSLY DEFINED.

A CALL WITH EITHER X OR Y = 999.0 WILL RESULT IN THE
PREVIOUSLY DEFINED VALUES OF X AND Y BEING USED.

DATA XSAVE/0.0/,YSAVE/0.0/

X1=X
IF (X1.EQ.999.0) X1=XSAVE
XSAVE=X1
Y1=Y
IF (Y1.EQ.999.0) Y1=YSAVE
YSAVE=Y1

FOR CALCOMP PEN PLOTTERS, USE

CALL PLOT(X1,Y1,-3)
CALL PLOT(X1,Y1,-3)

FOR CATHODE RAY TUBE PLOTTERS, USE

CALL CALCOMP(0.0,0.0,0.000,2)
CALL CALCOMP(X1,Y1,-6,3)

TO ADVANCE THE FILM, AND
TO RESET THE LOGICAL ORIGIN.

RETURN

END

KAVANS

SUBROUTINE KAXIS(X,Y,LABEL,NC,AL,ANGLE,AMIN,DA,DEL,ITURN)

SUBROUTINE KAXIS DRAWS A LABELED AXIS.

CALLING ARGUMENTS -

X = X-COORDINATE IN INCHES OF THE START OF THE ORIGIN. (REAL)
Y = Y-COORDINATE IN INCHES OF THE START OF THE ORIGIN. (REAL)
LABEL = ALPHANUMERIC TEXT TO BE PRINTED ALONG AXIS. (HOLLERITH)
NC = NUMBER OF CHARACTERS IN LABEL (INCLUDING ALL BLANKS).

A POSITIVE VALUE WILL PLACE THE TITLE ON THE COUNTERCLOCKWISE SIDE OF THE AXIS, USUAL Y-AXIS.
A NEGATIVE VALUE WILL PLACE THE LABEL ON THE CLOCKWISE SIDE OF THE AXIS, USUAL X-AXIS.
(INTEGER)

AL = LENGTH OF AXIS IN INCHES. (REAL)

ANGLE = ANGULAR ORIENTATION OF AXIS, DEGREES.

0 DEGREES = HORIZONTAL TO RIGHT, POSITIVE ANGLES ARE MEASURED COUNTERCLOCKWISE.
(REAL)

AMIN = MINIMUM VALUE OF VARIABLE ON AXIS.

VALUE RETURNED BY KALE. (REAL)

DA = SCALE FACTOR, UNITS/INCH, OF VARIABLE ALONG AXIS. (REAL)

DEL = UNITS/TIC MARK, RETURNED BY KALE. (REAL)

ITURN = NUMBER OF 90 DEGREE ROTATION OF NUMBERS INDICATING VALUE OF MAJOR TIC MARKS. (INTEGER)

COMMON/KACOM/NCPW, ICIN(2), ICOUT(1), INP(3), NGRAFS, NOGRAF, NCRT

DIMENSION LABEL(1), JFMT(12), IFMT(3)

DATA JFMT/4HF6.1,4HF6.2,4HF6.3,2HI2 ,4H,14X,4H2H10,2HI3 ,
12HI5,2HI6,2HI4,2HI3,2HI1/

HGTNUM = HEIGHT OF NUMBERS ALONG AXIS, INCHES.

DSTO = DISTANCE BETWEEN AXIS AND TOP OF NUMBERS, AND BETWEEN NUMBERS AND AXIS LABEL.

HGTSYM = MAXIMUM ALLOWABLE HEIGHT OF SYMBOLS IN AXIS TITLE.
INCHES.

DATA HGTNUM/0.100/, DSTO/0.09/, HGTSYM/0.130/

CENTER = NUMBER CENTERING VARIABLE.

PROFILE -

SOME SYSTEMS RIGHT JUSTIFY NUMBERS WRITTEN UNDER FORMAT STATEMENTS. OTHERS LEFT JUSTIFY THESE NUMBERS. THUS, ON SOME MACHINES THE MAJOR TIC MARK VALUES MAY BE SHIFTED EITHER LEFT OR RIGHT OF WHERE THEY SHOULD BE.

SOLUTION -

CENTER PERMITS CENTERING NUMBERS FOR PARTICULAR SYSTEM IN USE.

CENTER = 0.0, LEFT SIDE OF NUMBER WILL LINE UP WITH MAJOR TIC MARK

CENTER = 1.0, RIGHT SIDE OF NUMBER WILL LINE UP WITH MAJOR TIC MARK

CENTER = 0.5, MIDDLE OF NUMBER (AS FORMATTED) WILL LINE UP WITH MAJOR TIC MARK. THUS, WITH AN IS FORMAT, THE THIRD SPACE WILL LINE UP WITH THE TIC MARK.

DATA CENTER/0.5/

SIDE=1.0

IF (NC.LT.0) SIDE=-1.0

DST=DSTO

IF (SIDE.EQ.-1.0.AND.ITURN.EQ.0) DST=DST+HGTNUM

CIT=0.0

IF (ITURN.NE.0) CIT=-1.0

XOR=1.0

IF (SIDE.EQ.FLOAT(ITURN)) XOR=0.0

THETA=ANGLE*0.0174532952

SINT=SIN(THETA)

COST=COS(THETA)

T=ANGLE+FLOAT(ITURN)*90.0

KAXIS

```

TN=I*0.0174532952
CN=COS(TN)
SN=SIN(TN)

NUMBER ORIENTATION CORRECTIONS.

CON=SIDE*XOR*SINT*CI-CENTER*CONST*(CI+1.0)
CON1=SIDE*XOR*CONST*CI+CENTER*SINT*(CI+1.0)

PLOT MAJOR TIC MARKS.
R = RANGE OF VARIABLE.
NT = NUMBER OF TIC MARKS.

AMAX=AMIN+AL*DA
R=AMAX-AMIN
NT=INT(R/DEL+0.1)
XR=X
YR=Y

TIC HEIGHT = 0.1 INCHES.

XA=X+0.1*SIDE*SINT
YI=Y-0.1*SIDE*CONST

XC = CONSTANT INCREMENT ALONG X AXIS.
YC = CONSTANT INCREMENT ALONG Y AXIS.

YC=DEL*CONST/DA
YC=DEL*SINT/DA
CALL PLOT(XA,YA,3)
CALL PLOT(XA,YA,2)
J=1
NT4=NT
5 DO 10 I=1,NTM
  CALL PLOT(XB,YB,2)
  XR=XC+XC
  YR=YB+YC
  CALL PLOT(XB,YB,2)
  XA=XA+XC
  YA=YA+YC
10 CALL PLOT(XA,YA,2)

PLOT MINOR TIC MARKS, 10/INTERVAL.

NT4=10*NT
IF (J.EQ.2) GO TO 20
J=2
XA=XR+0.05*SIDE*SINT
YA=YI-0.05*SIDE*CONST
XC=-0.1*XC
YC=-0.1*YC
GO TO 5
20 XC=-10.0*XC
YC=-10.0*YC

WRITE VALUE AT TIC MARKS.

NT=NT+1
A=APS(DEL)
IF (A.LT.1.0.OR.A.GE.10000.0) GO TO 60

INTEGER FORMAT.

I=INT(AMIN)
J=INT(DEL)
IC=2
IP=5
IFMT(1)=JFMT(8)
IF (AMIN.NE.-10000.0) GO TO 24
IP=5
IFMT(1)=JFMT(9)
GO TO 25
24 R=APS(AMAX)

```

KAXIS

AD-A074 887

KAMAN AVIDYNE BURLINGTON MA

TRAP-ML-A TWO DIMENSIONAL THERMAL RESPONSE CODE TAILORED FOR TH--ETC(U)

NOV 78 W N LEE

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NL

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2 OF 2

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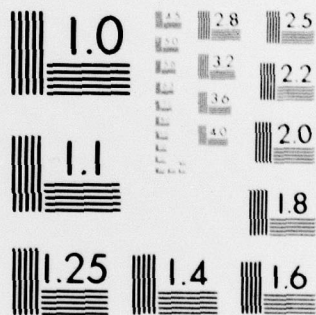


END

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MICROCOPY RESOLUTION TEST CHART
NATIONAL BUREAU OF STANDARDS-1963-A


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IF (R.LT.ABS(AMIN)) B=ABS(AMIN)
B=B+0.1
IF (R.GE.1000.) GO TO 25
IP=4
IFMT(1)=JFMT(10)
IF (R.GE.100.0) GO TO 25
IP=3
IFMT(1)=JFMT(11)
IF (AMIN.LT.0.0 .OR. DEL.LT.0.0) GO TO 25
IP=2
IFMT(1)=JFMT(4)
IF (P.GE.10.0) GO TO 25
IFMT(1)=JFMT(12)
IP=1
25 FIR=FLOAT(IR)*HGTNUM
XR=X-SIDE*DST*SINT+FIR*CON
YR=Y+SIDE*DST*COST-FIR*CON1
DO 30 I=1,NT
CALL KANUM(XB,Y3,HGTNUM,I,T,IFMT,IC,IR)
XR=XR+XC
YR=YR+YC
30 I=I+J
40 IF (NC.EQ.0) GO TO 45
LABEL AXIS.
S=FLOAT(NC)*SIDE+1.0
H=AL/S
IF (H.GT.HGTSYM) H=HGTSYM
TD=4*S
XR=X+0.5*(AL-TD)*COST-SIDE*SINT*(HGTNUM+2.0*DST+0.5*(1.0-SIDE)*H)
IF (ITURN.NE.0) XR=XR-(FIR+DST)*CTN
YR=Y+0.5*(AL-TD)*SINT+SIDE*COST*(HGTNUM+2.0*DST+0.5*(1.0-SIDE)*H)
IF (ITURN.NE.0) YR=YR-(FIR+DST)*STN
I=IABS(NC)
CALL KASYM(XB,Y3,H,LABEL,ANGLE,I,3)
45 DETJON
DECIMAL NOTATION.
50 B=A_0G10(A)
IF (P.GT.0.0) B=B+0.02
I=INT (B-0.01)
IF (I.GT.0.OR.I.LT.-2) GO TO 80
IFMT(1)=JFMT(3)
IF (P.LE.-1.0) IFMT(1)=JFMT(2)
IC=4
IP=5
A=AMIN
FIR=6.0*HGTNUM
XR=X-SIDE*DST*SINT+FIR*CON
YR=Y+SIDE*DST*COST-FIR*CON1
70 DO 72 I=1,NT
CALL KANUM(XB,Y3,HGTNUM,A,T,IFMT,IC,IR)
XR=XR+XC
YR=YR+YC
72 A=A+DEL
GO TO 40
EXPONENTIAL FORMAT.
80 A=10.0**T
AM=AMIN/A
ADA=DEL/A
NT=NT-1
IF (ADA.GE.0.01) GO TO 90
I=I-1
GO TO 80
90 IFMT(1)=JFMT(2)
IF (ADA.GE.1.0) IFMT(1)=JFMT(1)
IC=4
IP=5

```

```

      FIP=6.0*HGTNUM
      X0=X-SIDE*OST*SINT+FIR*CON
      Y0=Y+SIDE*OST*OST-FIR*CON1
100  DO 110 J=1,NT
      CALL KANUM(X0,Y0,HGTNUM,AM,T,IFMT,IC,IR)
      X0=X0+XC
      Y0=Y0+YC
110  AM=AM+ADA
      IFMT(1)=JFMT(2)
      IFMT(2)=JFMT(5)
      IFMT(3)=JFMT(6)
      IC=12
      IR=13
      X0=X0-0.4*CTN
      Y0=Y0-0.4*STN
115  CALL KANUM(X0,Y0,HGTNUM,AM,T,IFMT,IC,IR)
      INSERT EXPONENT.
      X0=X0+9.0*HGTNUM*CTN-HGTNUM*STN
      Y0=Y0+9.0*HGTNUM*STN+HGTNUM*CTN
      9.0 = 6 CHARACTERS IN NUMBER + 3 FOR X10.
      IFMT(1)=JFMT(4)
      IC=2
      IR=2
      IF (I.LE.0.AND.I.GE.-9) GO TO 117
      IR=3
      IFMT(1)=JFMT(7)
117  CALL KANUM (X0,Y0,HGTNUM,I,T,IFMT,IC,IR)
      GO TO 40
2    END

```

KAXIS

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